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EFFECTS OF STRAIN AMPLITUDE ON THE SHEAR MODULUS OF SOILS

Bobby O. Hardin

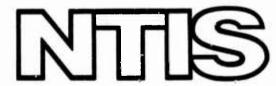
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Air Force Weapons Laboratory

March 1973

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TECHNICAL REPORT NO. AFWL-TR-72-201

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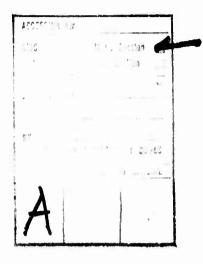


AIR FORCE WEAPONS LABORATORY

Air Force Systems Command
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FOREWORD

This report was prepared by the University of Kentucky, Lexington, Kentucky, under Contract F29601-72-C-0027. The research was performed under Program Element 63723F, Project 683M, Task 4A06.

Inclusive dates of research were 10 October 1972 through 2 November 1972. The report was submitted 9 January 1973 by the Air Force Weapons Laboratory Project Officer, Major Donald V. Harnage (DEZ).

The Contractor's report number is UKY TR63-72-CE23.

This technical report has been reviewed and is approved.

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Division

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ABSTRACT

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One hundred twenty-three simple shear tests of 24 different soils were conducted. Most were constant-amplitude repeated load tests. A few of the tests involved mixed amplitudes of loading with rest periods between loads. Based on the results, a practical procedure for reducing the shear modulus of soils with increasing strain amplitude was developed. It was shown that for a wide variety of soil types and conditions the procedure gives reasonably accurate results compared to values measured in the laboratory. The study of mixed amplitudes and rest periods indicated that the procedure can be applied to mixed traffic conditions.

TABLE OF CONTENTS

Secti	<u>on</u>	Pag									
I	INTRODUCTION	1									
	1. Pavement Evaluation	1									
	2. Review of Previous Report	2									
	3. Current Testing Program	2									
II	A PRACTICAL PROCEDURE FOR REDUCTION OF SHEAR MODULUS WITH INCREASING STRAIN LEVEL .	4									
	1. Objective	4									
	2. Definitions	4									
	3. Determination of the Reference Strain	6									
	4. Determination of the Shear Modulus	9									
	5. Examples of Use of the Procedure	18									
Ш	COMPARISON OF VALUES FROM THE PRACTICAL PROCEDURE TO EXPERIMENTAL DATA										
	1. Soils Tested	21									
	2. Values of R for Determination of Reference Strain	21									
	3. Comparison of Shear Modulus Values	34									
IV	MIXED LOADING AMPLITUDES AND REST PERIODS	46									
	1. Objective	46									
	2. Loading Programs and Recorded Stress-Strain Relations	46									
	3. Effect on the Shear Modulus	53									
v	CONCLUSIONS	56									
	Appendix I	59									
	Appendix II	61									

LIST OF FIGURES

Figure		Page
1	Schematic simple shear stress-strain relation.	5
2	Value of C_1 versus void ratio for sands with less than 15 percent fines.	8
3	Value of C_1 versus plasticity index, percent saturation, and void ratio for cohesive soils with more than 15 percent fines.	10
4	Reference strain versus value of \mathbf{C}_1 and maximum shear modulus.	11
5	Hyperbolic strain versus normalized strain for various numbers of cycles and rates of loading.	13
6	Hyperbolic strain versus normalized strain for various numbers of cycles and percents saturation, fast rate of loading, $T=0.01$, nonplastic soils with fines and low plasticity soils.	14
7	Hyperbolic strain versus normalized strain for various numbers of cycles and percents saturation, slow rate of loading, T = 10, nonplastic soils with fines and low plasticity soils.	15
8	Hyperbolic strain versus normalized strain for numbers of cycles, percents saturation, and rates of loading, high plasticity soils.	16
9	Normalized shear modulus versus hyperbolic strain.	17
10	Particle size distribution curves for nonplastic soils.	27
11	Particle size distribution curves for low plasticity soils.	28
12	Particle size distribution curves for high plasticity soils.	29
13	Comparison of the variation of experimental and calculated values of R with percent saturation,	

<u>Figure</u>		Page
	Air Force Silty Sand, Air Force Silty Clay, Vicksburg Loess, Vanceburg, Allen, Kentucky 55, Longhorn, and West Virginia Shale.	30
14	Comparison of the variation of experimental and calculated values of R with percent saturation, Six Kirtland Soils, Virginia Clay, Dover, Prestonsburg Sand, Ellsworth, Louisiana Clay, San Francisco Clay, Cheeks, and Nevada Clay.	31
15	Variation of normalized shear modulus with normalized strain as given by the practical procedure for various values of a.	35
16	Comparison of measured and calculated values of shear modulus, first cycle, WES Sand, St. John's Sand, and Air Force Silty Sand.	36
17	Comparison of measured and calculated values of shear modulus, first cycle, Air Force Silty Clay, Vicksburg Loess, and Vanceburg.	37
18	Comparison of measured and calculated values of shear modulus, first cycle, Allen, Kentucky 55, and Longhorn.	38
19	Comparison of measured and calculated values of shear modulus, first cycle, West Virginia Shale, Virginia Clay, and Dover.	39
20	Comparison of measured and calculated values of shear modulus, first cycle, Prestonsburg Sand, Kirtland #10-36, and Louisiana Clay.	40
21	Comparison of measured and calculated values of shear modulus, first cycle, San Francisco Clay, Ellsworth, and Cheeks.	41
22	Comparison of measured and calculated values of shear modulus, first cycle, Nevada Clay.	42
23	Comparison of measured and calculated values of	44

Figure		Page
24	Comparison of measured and calculated values of shear modulus, 100th cycle.	45
25	Loading Programs.	47
26	Recorded stress-strain relation, increasing load sequence, silty clay.	49
27	Recorded stress-strain relation, increasing load sequence, silty clay.	50
28	Recorded stress-strain relation, increasing load sequence, Vicksburg Loess.	51
29	Recorded stress-strain relation, decreasing load sequence	52
30	Normalized shear modulus versus normalized strain for silty clay, mixed amplitudes and rest periods.	54
31	Normalized shear modulus versus normalized strain for silty sand, mixed amplitudes and rest periods.	55
32	Typical Recorded Stress-Strain Relations.	62

LIST OF TABLES

<u>Table</u>		Page
1	Data for Simple Shear Tests of Clean Sands and other Sands with High Permeability.	22
2	Data for Simple Shear Tests of Nonplastic Silty Sands.	22
3	Data for Simple Shear Tests of Low Plasticity Soils.	24
4	Data for Simple Shear Tests of High Plasticity Soils.	26
5	Values of R for Sands with Less than 15 Percent Fines.	33

ABBREVIATIONS AND SYMBOLS

a = parameter in modified hyperbolic stress-strain relation

C₁ = parameter defined by equations 4 and 5

e = void ratio

e_{xp} = base of natural logarithms

F = function of void ratio defined by equation 2

G = shear modulus

G_{max} = maximum shear modulus

N = number of cycles

PI = plasticity index

R = parameter relating G_{max} and τ_{max}

S = percent saturation

T = time in minutes to reach a normalized strain equal to one

 γ = shear strain

 γ_h = hyperbolic strain

 $\gamma_{\mathbf{r}}$ = reference strain

 σ_{cham} = chamber pressure

σ_o = effective mean principal stress

 ϕ = effective angle of shearing resistance

 τ_{max} = maximum shear stress

SECTION 1

INTRODUCTION

1. Pavement Evaluation

This document reports research that is part of a larger Air Force
Weapons Laboratory project to develop a pavement evaluation procedure. The
following steps are involved in the pavement evaluation procedure, which for
the present uses linear elastic analysis for determination of the stresses and
strains in the pavement structure.

The first step is to measure the shear modulus for different layers of the pavement structure in-situ, using a nondestructive vibration testing method. The shear modulus thus measured will be for very small strain amplitudes, on the order of 10^{-5} in/in or less. Because the stress-strain relations for paving materials are nonlinear, the secant shear modulus for larger strain amplitudes produced by an aircraft loading will be smaller than the modulus measured by the nondestructive testing method.

The second step is to make the proper reduction in the measured shear modulus to correspond to the strain produced by an aircraft loading.

The third step is to use this shear modulus in a finite-element analysis for stresses and strains in pavement structure under load.

The fourth and final step is to assess the amount of damage or pavement distress that will be produced by the loading.

This report presents a procedure for the second step, the reduction of shear modulus with increasing strain level.

2. Review of Previous Report

The first phase in developing a procedure for the reduction of shear modulus with increasing strain level was to design and construct testing equipment capable of accurate measurement of the shear stress-strain relation for soils, over a wide range of strain levels. The range was from about 10⁻⁵ in/in to failure, i.e., after a given number of constant-amplitude cyclic loads each sample was loaded to failure. Also, a series of tests with two soils, a silty sand and a silty clay, were conducted to assess the relative effects of various parameters, such as density, percent saturation and confining stress, on the shear stress-strain relation for cyclic loading. Reference 1 reports on this phase of the research, and the testing equipment, its capabilities, and the testing procedure developed are described in detail. Examples of recorded stress-strain curves are given, and the methods of analysis of the data are discussed. The same equipment, procedures, and most of the methods of analysis were used for the current testing program. They will not be described in detail herein, since the information is available in reference 1. However, a brief description of methods and procedures is given in Appendix II. Some of the definitions presented in Section II are amplified in reference 1.

3. <u>Current Testing Program</u>

The objective of the current testing program was to determine whether

^{1.} Hardin, Bobby O., Characterization and Use of Shear Stress-Strain
Relations for Airfield Subgrade and Base Course Materials, Technical
Report No. AFWL-TR-71-60, Air Force Weapons Laboratory, Kirtland
AFB, NM, July 1971.

or not the characterization of the shear stress-strain relation presented in reference 1 is applicable to a wide variety of soils. A total of 80 specimens of 20 different soils were tested with differing parameters, such as soil type, strain amplitude, density, and percent saturation. Including Phase I, a total of 129 tests on 24 different soils have been conducted and analyzed. Data for only six tests appeared to be faulty (i. e., an error in conducting the test) and were thrown out. The relationships presented here are based on 123 tests of 24 different soils.

A procedure for reducing the shear modulus with increasing strain level, recommended for use in the Air Force pavement evaluation technique, is presented in Section II. In Section III values of the shear modulus determined by this procedure are compared to the experimental data in order to verify the procedure and to show the probable magnitude of error in using the procedure. A few of the tests involved more general loading histories, with mixed loading amplitudes and rest periods between loads. Results and discussions of these tests are presented in Section IV.

SECTION II

A PRACTICAL PROCEDURE FOR REDUCTION OF SHEAR MODULUS WITH INCREASING STRAIN LEVEL

1. Objective

The objective of this section is to present the procedure recommended for reduction of shear modulus with increasing strain level in a practical form, for use in pavement evaluation, unencumbered by details of testing or presentation of the supporting data.

2. Definitions

The following parameters are used and are defined with reference to figure 1:

Maximum Shear Modulus = G_{max} = the initial tangent modulus, or secant modulus for strain amplitude $\leq 10^{-5}$ in/in (For pavement evaluation this quantity is to be measured by the nondestructive vibratory test);

Maximum Shear Stress = τ_{\max} = strength of the specimen in simple shear, defines assymptote in figure 1 (This quantity has been related emperically to G_{\max} , see Appendix I. However, if measured values of τ_{\max} are used, a rate of loading corresponding to T=0.1 will suffice, eventhough the value of T for actual aircraft loading may be 0.01);

Reference Strain = $\gamma_r = \tau_{max}/G_{max}$, defined by the intersection of the initial tangent line and strength assymptote in figure 1, and emperically related to G_{max} in this section;

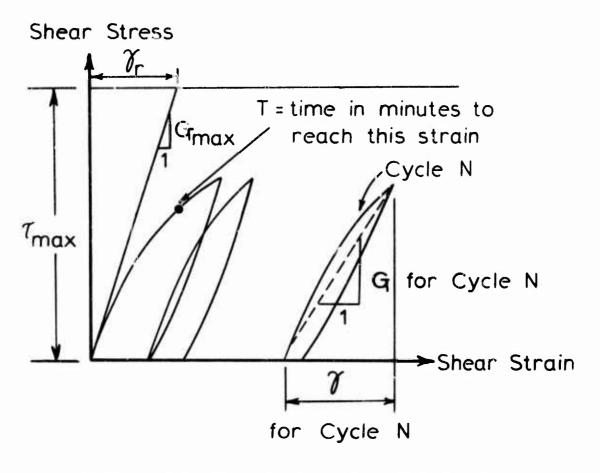


Figure 1. Schematic simple shear-stress strain relation.

Normalized Strain γ/γ_r , where γ is the shear strain;

Hyperbolic Strain $= \gamma_h$ a parameter defined by equation 9 (If equation 9 is substituted into equation 8, the relationship between normalized shear modulus and normalized strain is obtained. This relationship changes with the value of a, as shown in figure 15, depending on the values of S, N, T, and the type of soil; see equation 19.

An alternate method of presentation is used in this section.

Graphs based on equation 9 are presented giving the value of γ_h corresponding to a value of γ/γ_r , depending on the values of S, N, T, and soil type. With the value of γ_h determined, the normalized shear modulus is given by the simple hyperbolic equation 8.);

Shear Modulus = G = secant shear modulus for a given strain amplitude and cycle as shown in figure 1;

Normalized Shear Modulus = G/G_{max};

Number of Cycles = N;

Strain Time = T - time in minutes to reach a normalized strain equal to one;

Percent Saturation = S = ratio of volume of water to volume of voids in the soil, expressed as a percentage, and;

Void Ratio = e = ratio of volume of voids to volume of solids in the soil.

3. Determination of the Reference Strain

The reference strain is given by the following equation derived in

Appendix !.

$$\gamma_{\rm r} = \frac{G_{\rm max}}{F^2 R^2} \left[0.6 - 0.25 \text{ (PI)}^{0.6} \right]$$
 (1)

where PI = plasticity index, G_{max} is in psi,

$$\mathbf{F} = \frac{(2.973 - e)^2}{(1 + e)} \tag{2}$$

R = 1100, for sands with less than 15 percent fines and R = 1100 - 6S, for cohesive soils with more than 15 percent fines

Let

$$C_1 = \frac{F^2 R^2}{0.6 - 0.25 \text{ (PI)}^{0.6}} \tag{4}$$

Then for sands with less than 15 percent fines, PI = 0 and R = 1100 giving

$$C_1 = (2.017 \times 10^3) \text{ F}^2$$
 (5)

For cohesive soils with more than 15 percent fines, substituting $R=1100\,$ - 6S into equation 4

$$C_1 = \frac{F^2 (1100 - 6S)^2}{0.6 - 0.25 (PD)^0.6}$$
 (6)

Finally

$$\gamma_{r} = \frac{G_{\text{max}}}{C_{1}} \tag{7}$$

With Gmax in psi.

The v_i lue of C_1 for sands with less than 15 percent fines, as defined by equations 2 and 5, is given in figure 2. The value of void ratio is all that is needed to determine C_1 for this case. The value of C_1 for cohesive soils with more than 15 percent fines, as defined by equations 2 and 6, is given in figure 3.

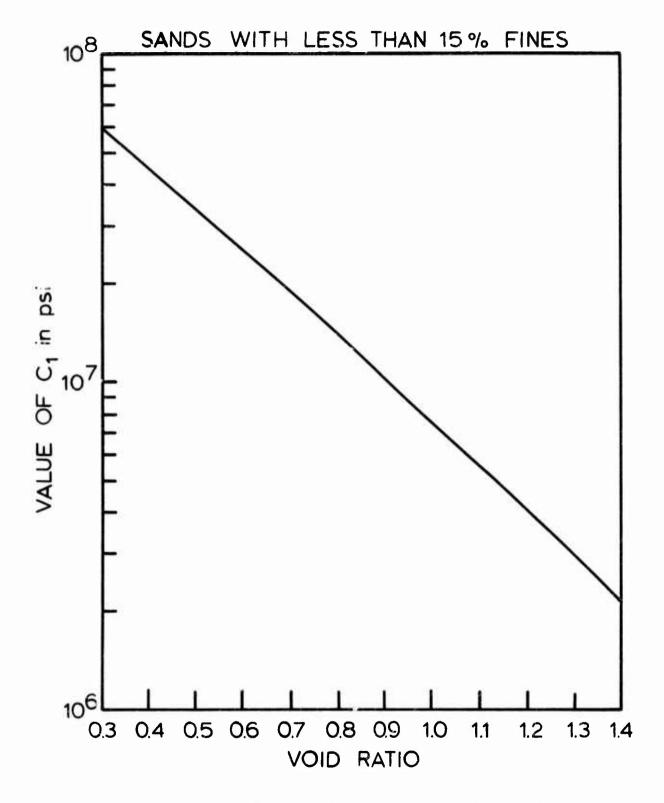


Figure 2. Value of C_1 versus void ratio for sands with less than 15 percent lines.

For this case the values of the plasticity index and percent saturation are needed in addition to the value of the void ratio. The dashed line in figure 3 shows how to use the figure. First, using the left hand side of the graph with the value of plasticity index as abscissa, determine an ordinate depending on percent saturation. Extend this ordinate to the right to determine an abscissa for the right hand side of the graph, corresponding to the value of void ratio. The second abscissa determines the value of C_1 . The dashed line represents the case where PI = 25, S = 50, and C = 1. The value of $C_1 = 4.9 \times 10^6 \, \mathrm{psi}$.

Having determined the value of C_1 from either figure 2 or figure 3, depending on the soil type, the reference strain can now be determined from figure 4. Using the value of C_1 as abscissa, the value of the reference strain is read as ordinate, from the curve corresponding to the value of G_{max} in psi. The dashed lines in figure 4 are for C_1 = 4.9 x 10⁶ psi and G_{max} = 8000 psi, γ_r = 16.5 x 10⁻⁴ in/in.

4. Determination of the Shear Modulus

The normalized shear modulus is given by

$$\frac{G}{G_{\text{max}}} = \frac{1}{1 + \gamma_{\text{h}}} \tag{8}$$

where

$$\gamma_{\rm h} = \frac{\gamma}{\gamma_{\rm r}} \left[1 + a \, e_{\rm xp} - \left[\frac{\gamma}{\gamma_{\rm r}} \right]^{0.4} \right]$$
 (9)

The value of a is defined by one of the following equations, depending on the type of soil

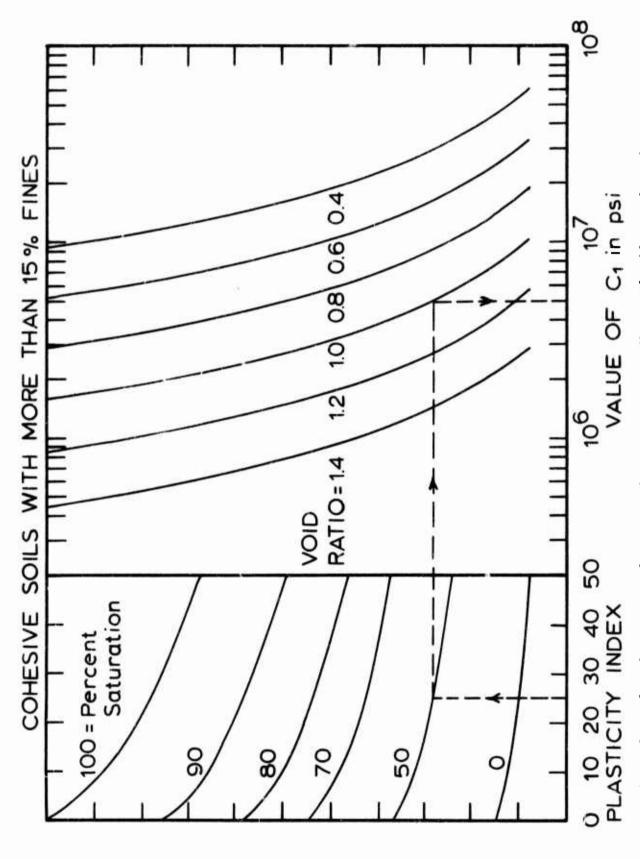


Figure 3. Value of $C_{\rm I}$ versus plasticity index, percent saturation, and void ratio for cohesive soils with more than 15 percent fines.

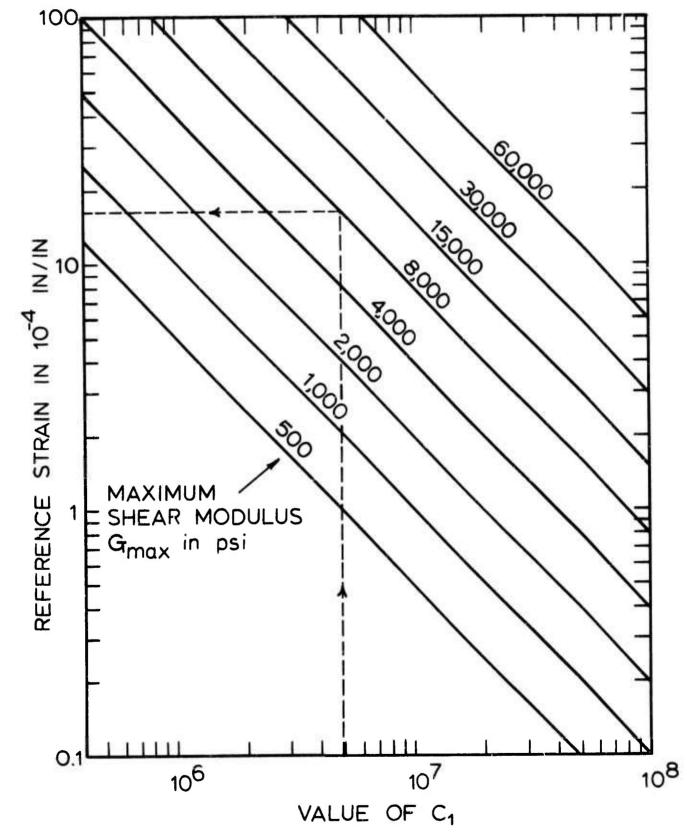


Figure 4. Reference strain versus value of \mathbf{C}_1 and maximum shear modulus.

$$\begin{bmatrix}
(3.85/N) - 0.85 \end{bmatrix} T^{0.025} & \text{for clean dry sands} \\
1.6(1 + 0.02S) T^{0.2}/N^{0.6} & \text{for nonplastic soils with fines and low plasticity soils} \\
0.2(1 + 0.02S) T^{0.75}/N^{0.15} & \text{for high plasticity soils with liquid limit > 50}
\end{bmatrix} (10)$$

The values of γ_h as defined by equations 9 and 10 are given in figures 5, 6, 7 and 8 for various cases. Figure 5 is for clean dry sands. The solid curves give the relationship between γ_h and γ/γ_r for a fast rate of loading, T = 0.01. The dashed curves are for a slow rate of loading, T = 10. Curves are shown for N = 1, 2, 10, > 100. The dashed and solid curves coincide for 10 and > 100 cycles. Having determined the reference strain according to the procedure given in paragraph 2, the normalized strain can be computed, and the hyperbolic strain determined from figure 5 for clean dry sands. Similar graphs are given in figures 6 and 7 for nonplastic soils with fines and low plasticity soils. In figures 6 and 7 the solid curves are for 60 percent saturation and the dashed curves are for 100 percent saturation. Figure 6 is for a fast rate of loading, T = 0.01, and figure 7 is for a slow rate of loading, T = 10. Both sets of curves in figures 6 and 7 approach the hyperbolic line for a large number of cycles. Again, knowing the value of reference strain, the value of the hyperbolic strain can be determined from these figures for a given value of strain. Figure 8 is a similar graph for high plasticity soils. For the fast rate of loading, T = 0.01, the hyperbolic line defines the relationship for all numbers of cycles and percents saturation. For the slow rate of loading, T = 10, the solid curves define the relationship for 60 percent saturation, and the dashed curves

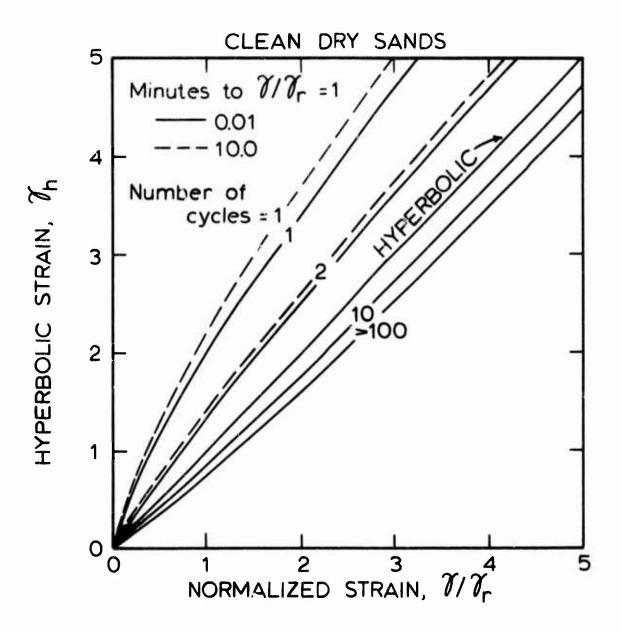


Figure 5. Hyperbolic strain versus normalized strain for various numbers of cycles and rates of loading.

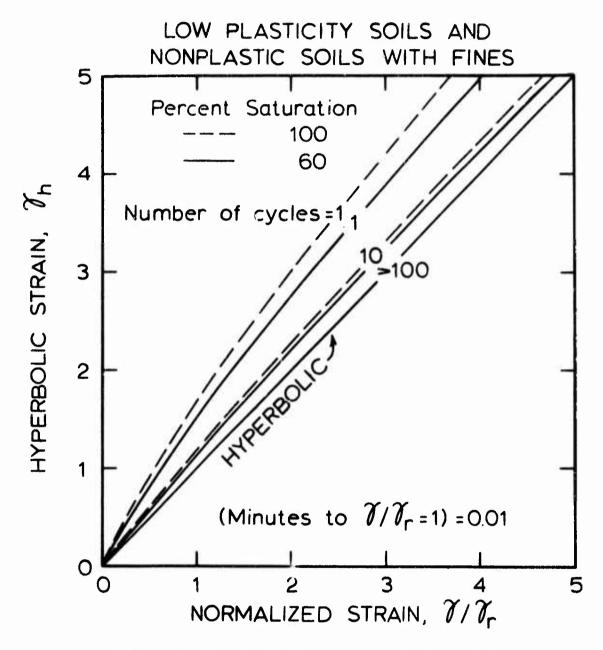


Figure 6. Hyperbolic strain versus normalized strain for various numbers of cycles and percents saturation, slow rate of loading, T = 0.01, nonplastic soils with fines and low plasticity soils.

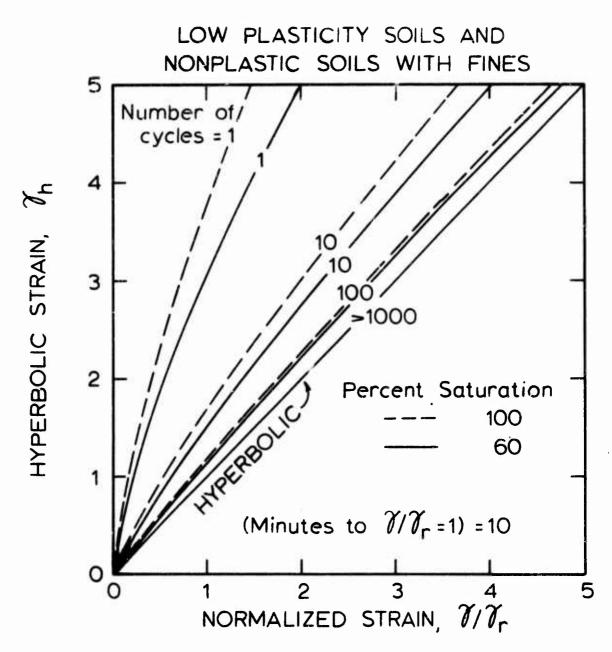


Figure 7. Hyperbolic strain versus normalized strain for various numbers of cycles and percents saturation, slow rate of loading, T = 10, nonplastic soils with fines and low plasticity soils.

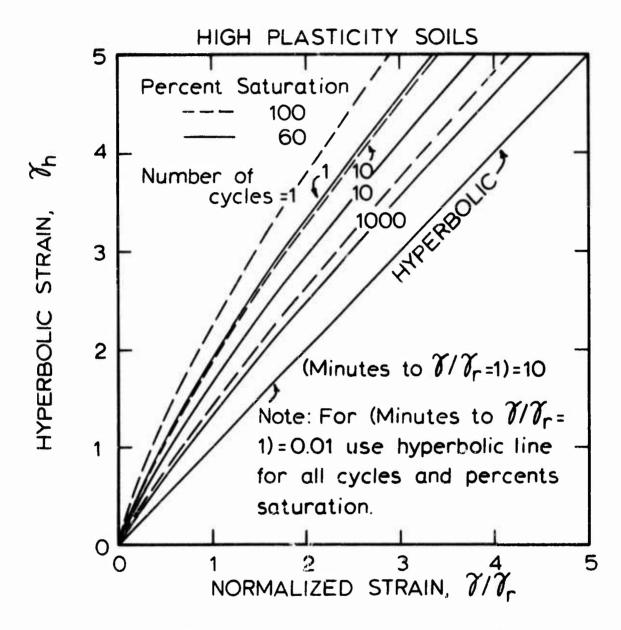


Figure 8. Hyperbolic strain versus normalized strain for numbers of cycles, percents saturation, and rates of loading, high plasticity soils.



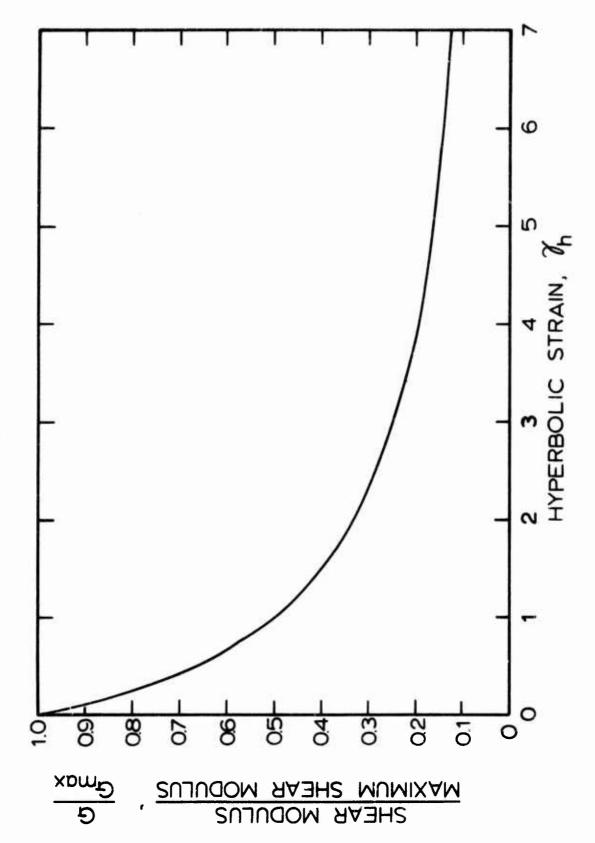


Figure 9. Normalized shear modulus versus hyperbolic strain.

are for 100 percent saturation. With the value of hyperbolic strain determined from one of figures 5, 6, 7, or 8, depending on soil type, the normalized shear modulus can be determined from figure 9.

5. Examples of Use of the Procedure

For pavement evaluation the type of soil (including PI and particle size) and values of e and S would have to be estimated from available knowledge of the subgrade or from a core sample; the value of G_{max} would be measured by nondestructive vibratory testing; the value of N would come from traffic records; the value of γ would be determined by the finite-element analysis, and; the value of T would depend on the speed of the aircraft.

Example number 1:

Given: A clean dry sand,

$$G_{\text{max}} = 18380 \text{ psi}$$
,

$$e = 0.62$$
,

$$N = 1$$
.

$$\gamma = 18.6 \times 10^{-4} \text{ in/in,}$$

T = 81 (a very slow rate of loading).

Find: The shear modulus, G;

From figure 2, using
$$e = 0.62$$
; $C_1 = 2.36 \times 10^7$ psi.

From figure 4, using
$$C_1 = 2.36 \times 10^7$$
 and $G_{max} = 18380$ psi;

$$\gamma_{r} = 7.5 \times 10^{-4} \text{ in/in.}$$

Calculate, using
$$\gamma = 18.6 \times 10^{-4}$$
 and $\gamma_r = 7.5 \times 10^{-4}$;

$$\gamma/\gamma_r = (18.6 \times 10^{-4}) / (7.5 \times 10^{-4}) = 2.48.$$

From figure 5, using $\gamma/\gamma_{\rm r}=2.48$, N = 1, and T = 81; $\gamma_{\rm h}=4.4$ From figure 9, using $\gamma_{\rm h}=4.4$; $G/G_{\rm max}=0.185$ Calculate, using $G/G_{\rm max}=0.185$ and $G_{\rm max}=18380$ psi; G=(0.185) (18380) = 3400 psi

The same parameters given in this example were measured for test no. 101 of the WES Sand (see figure 16). The measured shear modulus, G = 3320 psi, in this case, was less than 3 percent difference between calculated and measured values.

Example number 2:

Given: A low plasticity soil,

PI = 6,

Percent passing no. 200 sieve = 96,

 $G_{\text{max}} = 12680 \text{ psi},$

e = 0:67,

S = 73

N = 10,

 $\gamma = 12.9 \times 10^{-4} \text{ in/in,}$

T = 0.38 (a medium fast rate of loading).

Find: The shear modulus, G;

From figure 3, using PI = 6, S = 73, and e = 0.67;

 $C_1 = 8.0 \times 10^6 \text{ psi.}$

From figure 4, using $C_1 = 8.0 \times 10^6$ psi and $G_{max} = 12680$ psi;

$$\gamma_{r} = 15 \times 10^{-4} \text{ in/in.}$$

Calculate, using $\gamma = 12.9 \times 10^{-4} \text{ and } \gamma_{r} = 15 \times 10^{-4}$;

 $\gamma/\gamma_{r} = (12.9 \times 10^{-4}) / (15 \times 10^{-4}) = 0.860$.

Since $T = 0.38$, use the average between figures 6 and 7,

also using $\gamma/\gamma_{r} = 0.860$, $N = 10$, and $S = 73$;

 $\gamma_{h} = [(1.0 \text{ from figure 6}) + (1.44 \text{ from figure 7})] / 2 = 1.22$.

From figure 9, using $\gamma_{h} = 1.22$; $G/C_{max} = 0.450$

Calculate, using $G/G_{max} = 0.450$ and $G_{max} = 12680$ psi;

 $G = (0.450) (12680) = 5700$ psi

The same parameters given in this example were measured for test no. 27 of the Vicksburg Loess. The measured shear modulus G = 5580 psi, in this case, was less than 3 percent difference between calculated and measured values. Not all cases would compare this well, but these were not chosen specifically to show a good comparison.

SECTION III

COMPARISON OF VALUES FROM THE PRACTICAL PROCEDURE TO EXPERIMENTAL DATA

1. Soils Tested

Classification data for each of the soils tested and various test parameters for each of the tests are given in tables 1, 2, 3, and 4. In these tables the soils are grouped into four different categories. Table 1 gives data for clean sands and other sands with high permeability. Table 2 gives data for nonplastic silty sands. Table 3 gives data for low plasticity soils and table 4 for high plasticity soils. The particle size distribution curves for these soils are given in figures 10, 11, and 12. Those in figure 10 are for nonplastic soils, corresponding to the soils in tables 1 and 2. Figure 11 is for low plasticity soils, corresponding to table 3, and figure 12 is for high plasticity soils, corresponding to table 4. Study of these tables and figures will show that the testing program covered a wide variety of soil types and large ranges of test parameters.

2. Values of R for Determination of Reference Strain

Rearranging equation 16 in Appendix I gives

$$R = \frac{G_{\text{max}}}{F} \left[\frac{0.6 - 0.25 \, (PI)^{0.6}}{\tau_{\text{max}}} \right]^{1/2}$$
 (11)

For each test, using the measured values of PI, e, $\tau_{\rm max}$, and $G_{\rm max}$, a measured value of R can be calculated with equations 2 and 11. In figures 13 and 14 measured values of R for 22 of the soils tested are compared to values calculated with the second of equation 3, for cohesive soils with more than 15 percent fines. In general, equation 3 is a fairly good representation of the

Table 1. Data for Simple Shear Tests of Clean Sands and Other Sands with High Permeability.

					PERLENT	UF TESTS	NJ. 200 51	tvE = (
rest NO.	MECHRO NO.	OID	PERCENT SATURATION	CHAMBER PRESSURE KG/SU CM	PAINCIPAL STAESS DIFFERENCE	INSTIAL G	MECUVENY MAR & PS8	CYCLES BEFORE RECUVERY	SHEAR STRESS INCHEMENT	MAXIMUM SHEAR STRESS	SHEAR SIRESS RATIU	DAUJ BAN BAN BAN BAN BAN BAN BAN BAN BAN BAN
113	121	(3)	14)	151	KU/SU CM	171	4.44		KG/SG CM	KG/SU CH	4171	PER HOUR
34	121	0.05	0.	0.5			141	(9)	(10)	(11)	(1/1	57.903
35	:	0.66	٥.	1.0	0.127	19543.	10440.	1401	0.457	0.521	4.5297	14.014
36	;	3.48	ů.	3.5	0.147	12010.	12/30.	3480	3.161	0.570	0.2423	34. 834
17	i	0.49	o.	0.5	0.127	11100.	13013.	1430	0.343	0.516	0.03/4	40.236
34		0.67	ů.	1.0	0.115	10270.	19240.	645	150.0	0.415	U.6847	37.131
94	i	3.70	ű.	1.0	3.115	19643.	16444.	202	0.323	1.039	U. 310a	10.247
ñ 3	i	0.64	J.	1.0	0.000	15303.	20080.	94.15	3.462	0.913	0.5066	169.969
84	i	4.64	ű.	1.0	0.115	10040.	15640.	i	0.451	0.481	0.4596	0.222
101	i	9.62	0.	1.0	0.110	10300.	19170.	19	0.469	0.962	0.40/1	0.227
115		0.67	J.	1.0	0.067	19310.	17140.	320	0.462	0.909	0.50#7	12.247
					NUMBER PERCENT PLASTIC	INS SAND OF TESTS PASSING LITY INDEX LIMIT = N	NU. 200 51	EVE - 14	•			
NG.	PECUPA Nu.	OLTA #	PERCENT	CHAMMEN PRESSURE KG/3Q CM	PRINCIPAL STRESS DIFFERENCE KG/SQ LM	INITIAL MAR G PSI	MAR G PSI	#FCOVERY #FCOVERY	SHEAR STRESS INCREMENT NG/SU CM	MAXIMUM SHEAR STRESS RG/SQ CM	SHEAR STRESS RATIO	HATE KG/SQ EN PER HOUR
111	121	6 5 7	161	151	161	(7)	181	191	(10)	(11)	1121	1131
76	1	0.74	100.	0.5	0.096	1254).	11690.	4083	0.196	U-364	4.5389	22.201
17	l l	J. 5 L	130.	3.0	0.034	26160.	27460.	1114	0.548	2.511	0.21#2	24.148
19	ı.	0.65	AA.	1.5	0.076	£1490.	21760.	2155	0.211	1.610	0.1309	36.032
8.3		13.64		1.0		1403.1	17471.	310.	0.427	1 244	0. 1147	24 . G 64

Table 2. Data for Simple Shear Tests of Nonplastic Silty Sands.

					PERCENT	OF TESTS ASSIGN	NJ. 200 51	EVE - 44	•			
(1) 54	HELCHU NO.	VOS 0 VAT 10 E31 J=33 0.45	PERLENT SATURATION (4) 36.	CHAMBER PRESSURF KG/SJ CM (51 U.5 U.5	PHINCIPAL STRESS OIFFERFICE RG/SU UM (b) USUNDA OLLHU	INITIAL MAA U PSI 171 16#1J. 17290.	IICII. ICI ICI ICI AECIALNA	POR TOTA TOTA TOTAL WSCHAFHA MELOWE CACTEZ	SHEAR STRESS INCREMENT RU/SU CM ELUI O-438 U-225	MAXIMUM SMFAR STMESS KG/SQ CM (11) 1.004 0.914	SHEAR STRESS PATTU (12) 0.4344 0.2459	LOAU PATE RG/SU CH PER HUUN (13) 35-435
57 56 54	į	0.41 0.34 0.42	5d. 19. 59.	1.3	0.044	1977J. 228dJ. 1569J.	2261J. 2582J. 1839J.	1/60 646 811	U.425 0.428 U.538	1.734	0.3045	43.326 31.561 41.126
					NUMBER PERCENT PLASTIC	SBURG SAN OF TESTS OF SEST UNITED LIMIT + Y		EVE = 34	•			
HEST NO.	RFCCF5	VIII D	STURATION PERCENT	ENERGER PRESSURE ENERGER	PRINCIPAL STRESS JIFFERENCE RG/SU CM	INITIAL MAX G Pal	MAA G PSI	HECGVERY HEFTRE	SHEAR STRESS INCREMENT RG/SU CM	MAREMUM SHEAH STHESS KG/SU CM	SHEAR STRESS PATIO	EOAO HATE RG/SU CH PEH HOUR
111 72 73 74 75	1 1 1 1	(3) 0,59 0,60 3,93 3,94	1 + 1 1 7 . 1 5 . 4 7 . 4 9 .	15) 1.0 J.7	161 0.048 0.344 0.394 0.376	(7) 15463. 13713. 11220. 21653.	17150. 21781. 2333. 17190.	450 1590 7438 (4)	(10) 0.317 0.562 0.583 0.520	1.11 0.783 0.795 0.796 1.290	0.4044 0.5944 0.7306 0.4032	11.11

Table 2. Data for Simple Shear Tests of Nonplastic Silty Sands (cont').

					% () P () E M () E M (- 1	9 10 299 31 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	lc¥* 4 - ≱4				
1651	***	¥ 11 1 + 4 1 1;	204(EST 5413441104	CHRMULK PRE32746	PRINCIPAL	MAA	AECUIVERY	STELLS	SEHESS	MAKEMUM Set Ah	SHEAM	HATE
11)	(2)	(5)	161	151 40/53 (4	∪[#fixi6%c# Ku/34 € 4 (6)	173	141- 521	141	AU/SW EM	58HE35 HG/3G LM (111)	ELL)	PER MITTER ELIT
113	1	J.53	•5• •/•	0.5 1.0	J. J86	16213.	17702.	1964 196 974	U.293 U.555 U.756	U-627 U-955 U-851	0.4606 6.5408 3.3630	41.345
1.5	i	0.74	17:	1.7	J. UBA	14413.	10/5).	6/1	3.759	0.754	U. 3625	31.076
					MUMBER PERCENT PLASTIC	. 1441 - 1 16 - 16 - 16 16 - 16 - 16 16 - 16 -	41. 200 st	leve - 44	•			
86 , 8 76 Ju	Wie.	≠110 #ATT	AFORATION	CHANGER PRESSURF AD/SW CP	PRINCIPAL STRESS DIFFERENCE	HAR G	HAA U PSI	LTCLF5 HFFURE HFCUVEHT	SME BA STHE 3 2 INCHEMENT	46 3116 66 3116 76 3116	510 AF 5101 55 541 1 L	EUAU HAEL RG/SU EM
111	(2)	133	141	151	46/20 64	(1)	(4)	191	46/50 CM (EU) (459	111) 111) 1.927	1121	1133 14.346
104	l L	J.55	55.	1.0	0.087 0.087	19570.	23303.	2645	4.452	1.020	0.4431	14.181
110	1	3.71	43.	1.0	0.047	12840.	14444.	763	3.557	0.952	U.5448	+1.147
					MUMBEH PERCENT PLASTIC	IU NO. 4-1 UP TESTS PASSING LIV INDED LIMIT N	* 4 Yu. 2JU 51 C * NP	ltvt • le				
TEST NO.	HECTOR'S Note	VUEU MATEU	PERCENT	CHAMBLE PRESSURE KG/SW CM	PRINCIPAL STRESS DIFFERENCE	INITIAL MAG U Pai	HECOJERY G AAN B 29	MELUVEKY CYLLES	SHEAM STRESS INCREMENT	MAXIMUM SMFAH STHESS	SHEAR STHESS HATEU	LADJ HATE HU/SJ CM
111	(2)	111	14)	1.0	RG/Sy C4 163 0.387	171	(#)	[4]	46/50 C4 (Ly) U.345	46/50 CM (11) 0.866	4121	1641 6641 921-50
109	i	J.55	44.	1.5	J. UB?	14063.	14060-	104	0.407	0.441	0.4324	4.311
411	ι	3.53	30.	1.0	0.087	13067.	Lion.	934	J.411	0.020	J.4984	26.439
					NUMBER PERCENT PLASTIC	10 MD, 4-1 (# TESTS ! PASSING (IFV IMUE) LIMIT + N	40. 200 SI	1646 - 74	•			
TEST NU.	MC.	4010 44711	PERCENT	CHAMBER PPESSURE RG/SJ CM	PRINCIPAL STRESS DIFFERFNCE RG/SD LM	INITIAL MAR G Pai	HECOVERY HAN G PSI	MECHAENA MECHAENA CACTER	SHEAR STRESS INCHEMENT RG/SU CM	MARIMUM SHEAR STHESS RG/S4 CM	SHEAR STHESS HATED	LUAU RATE RU/SI CM PEH HUUR
113	(2)	131	(4) 70.	1.0	0.007	471 21963.	101 20050.	(9) 1249	0.471	1.019	0.1043	16.111
114	•	3.5%	**.	1.0	0.047	20410.	410 9 J.	474	3.531	0.753	0.7047	14.576
					MUMMER PERCENT PLASTIC	III MUL 7-1 UF TESTS I PASSIMŪ LITY INIER LIMIT + N	 	EVE - 45	•			
TEST No.	MECHAN ML.	44111	PEHLENI SATURATION	WPL25 CH BRE22NdF CHM40f V	PHENCIPAL STRESS OIFFEHENCE RG/SU CM	INITIAL MAK G PSI	RECUVERY MAR U PSI	MECOAFAA PALUKE PALEZ	SHEAR STHESS INCREMENT RG/SQ CM	MAXIMUM SHEAR STHESS RG/SW CM	SMEAR STRESS PATIO	LUAJ RATE RG/SQ CM PER HJUR
115	(2)	133	(4) 3 * •	1.3	6.04/	(7) 1056J.	13630.	193	4143	6.414	1121	1131
110	ı	Jens	34.	1.5	J. u#7	14673.	15037.	1972	0.503	1.204	0.4154	73.444
					NUMMER PERCENT PLASTIC Cluuto	114 1 4 35 E	*). 200 SI					
TEST lait.	MECCHU No.	- 4f [1]	PC4CENE SATURATION	ίμ∦ψηξα Ρατββυατ «G/» » ί μ	PRINCIPAL SIRESS DIFFERENCE RS/SW CM	INITIAL MAA G PSE	MEC JVC 4Y MAA J PSI	DEFUNE DECUVERY	SHEAR STHESS INCREMENT RUSSU CH	KC/20 CM STRE35 WAXIMUM	SHFAR STHESS HATLO	EUAD PAFE RU/SJ CM PER HUUM
117	(2)	(1)	1+1 32+	1-4	(a) J. 387	101-3.	4350.	191 2558	110)	9.842	0.3673	52.463
115	1	3.55	11.	2.3	J. J#7	. 4413.	10134.	334	3.300	1.030	U.1891	2.422

Table 3. Data for Simple Shear Tests of Low Plasticity Soils.

Alk Alect Still (Ale)
NUMBER (A. 15-15) - 17
REFURIT MAJER (A. 15-15) - 23
REATTION (ALE CONTROL ALE
11.51		1111	PLACENT	CHAMBER	PHINCIPAL	PALLIAL	46 - 341 47	CYCLES	SHE AH	MAALMIM	SHE AH	LUAU
	** **	- A (1 .	ATJEATION	PRESSINE	STAESS	MAK .	MAA .	otf let	STRESS	SHE AR	STALSS	HAIL
				4 J/SU CM	HEFERENCE	621	Pal	A E CHIVE HY	INCREMENT	STALSS	HATLO	AU/SJ LH
					AU/ SU (M				46/56 CM	AG/SW LM		PER IBIUM
4 4 3	124	6 5 3	4 4 5	(5)	163	171	1 11	6 +1	1101	1111	1121	4 4 5 3
	,	2.5 .	45.	3.5	1.125	4750.	10143.	10	U. U 5H	4.571	U. 000 1	0.269
	5	3.54	45.	J. 7	4-125	11-13.	16743.	21	0.100	4.571	U. 1843	0.760
	4	1.50	45.	J. 5	0.125	12740.	1 4 403.	7	U. 3UH	0.575	0.5312	4./45
- 1	O .	3.50	45.	1.0	J. 113	20773.	17930.	13	0.393	U.neż	U. 058 i	0.174
	7	3.56	45.	1.0	0.113	16534.	10733.	20	0.159	0.062	U. 1846	0.157
₹ .	+	3.50	45.	1.0	0.115	insou.	14100.	45	0.433	U. 86/	0.5027	0.111
,	ı	3.57	41.	4.5	4.14>	10453.	12000.	47	0-10-	0.775	U-1006	0.142
	2	4.57	41.	J. 5	0.125	12110-	12143.	17	U. 305	0.5/3	U.5783	4.123
,	,	3.57	41.	1.0	J. 113	16433.	1/313.	24	0.050	J. 862	U-0542	0.167
3	4	3.57	41.	1.3	0.113	16330.	16973.	14	0.154	0.862	0.1784	0.144
3	5	0.47	41.	1.0	U.113	17040.	17103.	14	0.417	0.862	4.4837	0.314
	52	0.57	41.	1.3	0.113	11940.	17223.	16	0.444	0. 462	0.4914	U. 319
•	à.	3.27	33.	V.5	0.125	8350.	10233.	17	0.312	0.450	0.6926	0.235
•	3	3.47	33.	1.0	0.113	13600.	14150.	23	J. 048	0.743	U.U655	0.364
4	4	J.67	33.	1.0	3.113	13203.	13320.	13	0.426	U. 740	0.5756	U. 319
5	ı	0.57	42.	0.5	0.125	4625.	12073.	626	3.307	0.575	3.5363	24.155
5	2	4.57	42.	1.0	0.111	14333.	14300.	515	U.U48	U. 862	0.0560	38.139
5	3	1.57	+2.	1.0	0.113	14330.	14510.	120	0.151	0.862	U. 1748	34.457
5	4	3.78	42.	1.3	0.113	14113.	14110.	545	U.424	0.862	0.4421	32.736
6	1	3.57	39.	1.0	3.413	13363.	15153.	6 50	3.418	0.863	U.5097	33.642
7	1	3.51	55.	0.5	0.125	iledu.	15740.	100	0.510	0.617	0.8004	19.148
	ı.	3,53	52.	1.0	0.113	15160.	15920.	800	0.611	0.443	0.6476	44.424
9	i i	3.46	0.00	U.5	0-125	12500.	13430.	1201	0.374	0.876	0.5701	42.543
10	1	3.44	62.	J. 5	0.125	11113.	1 1533.	2167	0.210	0.671	0.3127	42.713
11	L	3.51	55.	U.5	0.125	10410.	12640.	1130	J. 196	0.669	0.1505	20.172
14	1	3.50	57.	1.0	0.113	15020.	10 100.	945	0.217	0.995	0.2162	49.493
13	1	0.52	59.	V.5	0.125	11344.	13500.	120	0.518	0.707	6.7327	25.801
1 4	1	3.44	95.	4.5	9.398	4170.	4174.	4	0.289	0.269	1.0010	41.059
15	ı	3.45	100.	0.5	U. 098	4010.	0553.	1011	0.010	0.406	0.0251	4.100
15	2	J. 4>	130.	3.5	3.04#	7433.	7430.	1242	J. 107	U. 463	U.2307	24.605
15	3	3.45	103.	0.5	0. 548	1443.	4470.	1077	0.233	0.513	0.4542	40.512
40	1	4.19	96.	1.0	0.115	14910.	19930.	10	VAR	1.238	0.0000	VAR
40	2	0.34	76.	2.0	7.957	24134.	24130.	9	VAR	2.475	0.0000	PAY
46	1	J. 40	74.	0.5	0.126		3023.	84	VAH	0.909	0.0000	VAH
49	1	3.41	87.	1.0	3.394	15574.	14630.	90	WAR	1.002	0.0000	VAH

AIM FUNCE SILTY LUAY
MUMBER OF TESTS = 16
PERCENT PASSING MUS. 200 STEVE = 86.
PLASTICITY INFOR = 15.
LIQUID LIMIT = 36.

1651	BE CURY	4010	PERCENT	CHAMBER	PRINCIPAL	INITEAL	RECOVERA	CACFF	SHEAR	MALIMUM	SHE AH	LUAD
NU.	40.	HATTU	SATURATION	RESSURE	STRESS UIFFERENCE KG/SU CH	MAR G	PSI V	#FENHF	STRESS INCREMENT KG/SO CH	SHEAR STRESS KG/SU CM	STRESS	HATE KG/SQ EM PER HOUR
111	(2)	(3)	(4)	1 > 1	101-	171	1 0 3	(9)	(10)	(11)	1121	1111
16	117	J. 12	21.	0.5	0.098	71 10.	6.154.	849	0.014	U. 651	U.UI66	7.182
le	;	3.72	57.	3.5	0. 196	1473.	8-50.	1668	0.101	0.851	0.2124	41.866
16		3.72	57.	0.5	0.090	8960.	8440.	460	J.341	0.851	U. 0300	48,366
17	í	3.72	55.	J. 5	3.345	7740.	87500	2004	0.313	0.831	U. 3762	56-749
ii		3.72	54.	0.5	J. 09#	7142.	7100.	85	U. 026	0.743	4. 8209	54,428
19	i	0.72	55.	1.0	0.047	9143.	11740.	910	0.620	1.140	U.5440	55.290
20	i	3.67	40.	3.5	U. 19d	5330.	6773.	1015	0.154	0.011	U.1470	40.285
21	•	2.03	¥1.	J. 5	3.098	5230.	5230.	131	0.410	9.752	0.5273	49,190
22	•	3.74	100.	0.5	0.095	1040.	1990.	12	0.149	0.220	U.5856	32.4.52
23	i	3.72	78.	0.5	3.09#	2750.	4.160.	3410	0.073	0.437	0.1804	27.242
24	•	3.46	₹6.	1.0	U. 387	4924.	4210.	10	0.343	0.502	0.6427	64.592
25	:	3.74	61.	3.5	3.394	6290.	6210.	949	0.522	9.700	0.4821	45.494
	•				0.115	8943.	6940.	63	VAR	9.395	6.0000	VAR
41		3.67		0.5	0.126	11840.	10340.	*;	VAR	1.004	0.0000	PAY .
+3		0.65	•1.	0.5								
45		3.69 .	>6.	1.0	0.115	12413.	10040-	68	VAR	1.262	6.0000	MAR
48	1	3.07	60.	0.5	0.084	9230.	9230.	74	PAP	1.000	0.0000	VAR
4.0	2	3.67		1.0	0.084	1175J.	4000.	52	MAR	1.215	0.0000	PAY
90	J	J.68	54.	4.5	0.048	1240.	4320-	133	YAR	0.857	0.0000	PAY

VICKSUUMG LGESS
NUMBER OF TESTS = 9
PERCLNI PASSING NO. 230 SIFVE = 96.
PLASTICITY INDEX = 6.
LIJUIG LIMIT = 29.

1651 NO.	RECOMO NO.	V350	PERCENT	LHAMBER PRESSURE	PRINCIPAL STRESS	INTTIAL MAK G	RECOVERY	CYCLES	SHEAM	MARINUM SHEAR	SHEAR	LOAU
40.	40.		3410.411	KG/SU CM	UIFFERENCE	PSI	PSI	MECUVERY	INCHEMENT	STRESS	RATIU	KG/SQ CM
					KG/SU CM				KG/SU CM	RG/SQ CM		PER HOUR
111	(2)	13.	141	15)	(4)	171	LAS	193	(10)	(11)	1151	(13)
26	1	3.66	75.	3.7	4.041	4433.	3+33.	1966	0.208	0.755	0.2755	34.224
27	i	3.67	71.	1.4	6.077	12130.	1 # 800 .	1155	U. 504	1.283	C. 3934	47,444
28	i	3.50	66.	2.1	0.042	14130.	22500.	1313	U.55a	1.930	U.2871	46.751
2 4	i	3.65	73.	2.4	J. 040	16833.	17453.	5300	0.561	2.250	0.2493	454.229
30	i	0.04	61.	3.7	U. 394	#443.	8730.	2256	v.103	U. 840	4.1754	17.748
31	ī	J.44	11.	3.7	0. 494	9253.	11190.	1195	0.368	0.035	0.4572	36.238
34	i	0.44	70.	1.4	3.376	100 10.	14/33.	#355	0.314	1.370	0.2290	74.132
11	ĭ	0.06	76.	2.1	0.063	16720.	10723.	2770	0.110	1.774	0.1760	73.409
4.2	ī	J. u A	70.	J.5	U. U96	9430.	4433.	94	VAR	0.753	6.0000	VAR
• ?	1	0.0	r # *	v.,	0.040	,,,,,,						

Table 3. Data for Simple Shear Tests of Low Plasticity Soils (cont').

					PERCENT	(F. FESTs	NJ. 211 51	cVL + V4	•			
HSI No.	MECC. #5	4 1 E E E E	Property Pullsquiser	KINZA CM NHESSIAL CHVATEI	PHINCIPAL STRENS UIFFERENCE RUZSU CM	INITIAL MAA u Pal	RECURRENT NAP PSI	CYLLES BEFUNE SECUVERY	SHEAH STRESS INCHEMENT RG/SQ CM	MAREMUM SHEAN SERESS RG/SU CM	SHFAR STRESS HATEU	LITATE HATE RU/SI CH PER INDUS
111 64 55 56 67	1 1 1	(3) 3.67 3.69 3.69 3.69	1+! 6d. A7. VI.	151 0.5 0.7 1.4 0.7	161 0.098 0.394 0.074 0.098	1/1 12030. 13490. 14290.	181 1×160. 1>>du. 1>320. 1>070.	1030 1400 440 480 141	1101 0.374 0.490 0.591 0.352	(11) 0.mal 1.u44 0.445 0.947	1121 U.4391 U.4476 U.6278 U.3734	(13) 40. dem 40. be5 40. de5 39. 169
					PERCENT	LEGAL VIL.	NO. 200 51	Evt - 39				
FEST NG.	HECCHO NC.	¥1110 44ff0	PERCENT SATURATEUN	CHAMBER PRESSURE KG/SJ CM	PRINCIPAL STRESS DIFFERENCE	INITIAL MAA G PS1	RECUVERY MAK G PSI	CYCLES MECOVERY	SHEAR STRESS INCREMENT	MAKEMUM SHEAM STRESS	SHEAR STRESS PATIO	EHAII HAIE RG/SH CH
(1) #1 #4 #4	1 1 1	(3) 0.64 0.78 0.70 0.46	(4) 44. 51. 75.	151 0.5 0.5 1.0 1.5	46/50 CM (6) 0.09d 0.08d 0.08d	171 110JJ. 9260. 13230. 2129J.	(+1 11500. 7260. 14300. 25310.	1012	#G/\$\\\\ 1101 0.432 0.433 0.213 0.280	#6/50 C4 1111 0-617 0-504 0-967 1-102	(12) 0.6/02 0.6593 0.2169 0.2150	PER HOUR (11) +0.416 +1.318 41.305 38.137
					PERCENT	CF TESTS	NJ. 2JU 51	tvt • sa				
TEST NU.	MESCHO NG.	C1149	PERCENT SAFURATION	CHAMBER PRESSURF RG/SJ CM	PRINCIPAL STRESS DIFFERENCE	INITIAL MAA G PST	PECJVERY V RAM 129	C TCLES BEFORE RECOVERY	SHEAR STRESS INCREMENT	MAXIMUM SHEAN STRESS	SHEAR STRESS FATTU	LUAU RATE RG/SU CM
(1) 6t 10	(2) 1 1	(3) U.56 J.60	(4) 99. Luu.	157	0.048 10) 46\27 C4	6057° 1139°	4131. 4130. (41	193 2533 1274	0.551 0.553 (10) KC\20 CM	1111 1.020 0.633	(12) 0.2190 0.3444	PER HOUN 1131 101.06 7.794
					PLATFIC	UF TESTS	NJ. 200 51	f¥t + 46				
rest Mir.	*FC/380	4010 44710	PERCENT SATURATEUN	KONDO CM PHEPPOHE CHAMBEN	PHINCIPAL STRESS DIFFERENCE RG/SW-LM	INITIAL MAR u Poi	MECJYERY MAR U Pali	RFCUAERA GELONF CACFF?	SHEAR STRESS INCREMENT KG/SU CM	MAXIMUM SHEAR STRESS KG/SU CH	SHEAR STRESS RATE:	NEW THISTORY WATE FUAD
91 92 93	1 1 1	(3) (1.15 (1.43 (1.43	14) 14. 81. 70. 70.	(5) 1.5 1.0 1.5 0.5	(6) 0.094 0.044 0.076 0.094	17) 12220. 16910. 16100. 12150.	141 1974u. 1954u. 1864u.	191 1235 1847 1931	0.212 0.212 0.200 0.530 0.247	111) 0.691 1.312 1.731 0.769	(121 0.3065 0.2148 0.3144 0.3136	11 11 10 - 6 73 36 - 4 10 40 - 4 74 14 - 14 5
					MUMUEN PERCENT	ITY INDLA	43. 230 51	cVE = 85				
TEST Nu.	HELUHI) NI).	V019	PERCENT SATURATION	CHAMBER PRESSURE KU/SQ CM	PRINCIPAL STRESS DIFFERENCE	INITIAL MAR G PS1	MAK G MAK G MSI	SECOVERY LACTOR	SHEAR STRESS INCREMENT	MAXIMUM SHEAR STRLSS	SMEAN STRESS HATLU	EUAD RAFE KG/Su CM
95 96 97 98	(2) i 1 1 i	0.42 3.41 3.41 3.38	14) 04. 04. 98.	(5) 1.5 1.5 1.5	46/50 CM (6) (J.09m (J.084 0.076 (J.086	17; 9863. 11740. 13163. 10143.	10133. 10033. 10033. 10333.	19) 1683 1787 546 1114	#G/SQ C# 1101 0.35# 0.615 0.614 0.44#	46/54 CM 1111 U.847 L.113 1.518 1.285	1121 0.4228 0.5545 0.4075 0.3485	PER HOUR (13) >1.726 30.424 30.570 38.198
					PERCENT	DF TESTS PASSING TTY THOUS	NJ. 2JJ 51	EAE = 48	•			
TEST NU.	HFCURU NU.	VOI J 447 (-)	PFACENT SSTURATION	CHAMBER PHESSURE KIN/30 CM	PPINCIPAE SIMESS JIFFERENCE		46CUVEHV MAA U PSI	MECDAEKA JELOME CACFE?	SHEAR STRESS INCREMENT	MAKIMUM SHEAR STKESS	SHEAR STRESS HATIU	EOAD RATE RU/Sa CM
119 120 21 21	(2) : ! !	(3) 3,77 3,83 3,42 3,70	1 = 1 3 + . 7 3 . + 0 . 9 2 .	(5) 1.0 J.5 L.5	46/50 cm f51 0.007 0.007 0.047 0.047	171 11213. 14640. 16073.	(1) (17), (0)50, (1)7AJ, (1)1AJ,	141 1524 2301 203 430	RU/SU CM (10) U-020 U-211 U-214 O-332	RG/SQ C4 111) 1.073 1.094 1.920	(12) 0.3907 0.1926 0.1114 0.1617	PER HOUN (111 60, U53 32,503 1.629 12.234

Table 4. Data for Simple Shear Tests of High Plasticity Soils.

					PEHCENT	UF TESTS	NO. 200 ST	EA5 = A1				
						LIMIT .	64.					
16 ST (41) •	RECURD NO.	POID	PERCENT SATUKATION	CHANGER PRESSURE KG/SQ CM	PHINCIPAL STRESS DIFFERENCE AG/SQ CM	INITIAL NAA G PSI	HELOVEHY MAX G PSI	CYCLES BEFORE HECUVERY	SHEAR STRESS INCREMENT KG/SU CM	MAXIMUM SMEAR STRESS KG/SQ CM	SMEAN STRESS RATEG	LIJAD HATE KG/SQ CM PEH HOUR
2.4.2	(2)	(3)	143	151	161	173	181	191	1101	(11)	1121	61 11
51	ì	0.68	66.	1.3	0.099	10030.	10900.	33U 785	0.205	1.006	0.2042	42.37d 35.811
5 5	į	0.84	35.	3.5	0.098	10310.	10803.	11/4	0.425	1.086	0.3910	37.416
55 7s	1	0.61	72. 92.	1.0	0.084	2300.	1356J. 586J.	6976	0.534	6.355	0.2823	21.199
					PERCENT PLASTIC	OF TESTS PASSING ITY INDE: LIMIT =	NO. 200 51	EVE - 84	•			
TEST MO.	46C7#D 4C.	VOID HATEU	PERCENT SATURATION	CHAMBER FRESSURF RG/SQ CM	PRINCIPAL STRESS DIFFERENCE KG/SU CM	MAX G PSI	MAX G PS1	RECOVERA CACTER	SHEAR STRESS INCREMENT KG/SQ CM	MAXIMUM SHEAR STRESS NG/SQ CM	SHEAR STRESS RATIO	HATE RG/SU CH PEH HUUR
(1)	(2)	(3)	14)	(5)	161	171	(4)	(4)	1101	(11)	1121	1133
60	i i	1.20	57.	0.5	0.098	12710.	17640.	1111	0.314	1.100	0.2657	10.362
61	i	1.15	61.	1.3	0.098	14510.	17140.	603	0.544	1.003	0.3671	JW.196 34.604
• 3	į.	0.99	94.	0.5	U. 078	9190.	12450.	1160	0.351	0.636	0.5536	14.473
					PERC ENT PLASTIC	OF TESTS	NO. 200 ST	eve - 100				
NU.	N/J.	PA VIII	PERCENT	CHAMBER PRESSURE HG/SQ CM	PRINCIPAL STRESS DIFFERENCE KG/SQ CM	INITIAL MAX G PSI	RECOVERY MAX G PSI	RECUAERA CACTE?	SHEAR STHESS INCREMENT KG/SG CM	MAXIMUM SHEAR STRESS KG/SQ CM	NATIO STRESS SHEAR	HATE KG/SQ CM PER HOUR
649	121	(3)	141	151	663	(7)	10)	(4)	(10)	6111	(12)	113)
64	i i	1.42	02. 02.	1.0	J.098	7/50.	9830.	2209 940	0.214	1.172	0.2268	40.424
70	ì	1.45	99.	0.5	0.094	4940.	3690.	1357	U. 365	U.727	0.5020	39.455
71	1	1.39	97.	1.0	0.000	6300.	7700.	5177	0.340	0.905	0.3024	36.965
					PERCENT	OF TESTS PASSING LITY INDEX	NU. 200 SI	EVE • VA				
NO.	4FC0H0	OLTAR	PERCENT SATURATION	CHAMBER PRESSURE KG/SQ CM	PHINCIPAL STRESS DIFFERENCE KG/SG CM	INITIAL MAX G PSI	RECOVERY MAK G PS1	RECOVER A RECOVER A	SMEAR STRESS INCREMENT RG/SQ CM	MAXIMUM SHEAR STRESS KG/SQ CM	SMEAH STRESS RATIO	RATE KL/SQ CM PER HOUR
(1) 125	(5)	(3)	161	151	0.067	171	181	(9) 324	1101	(11)	0.2689	1.120
					NUMBER PERCENT	ITY INDE	NO. 200 SI	LVE	•			
TEST Mil.	**CU90	44110 4010	PERCENT SATURATION	CHAMBER PRESSURE KG/SU CM	PAINCIPAL STRESS DIFFERENCE AG/SG CP	INITIAL MAX G PSI	RECOVERY MAX G PS1	CYCLES BEFURE RECOVERY	SHEAR STRESS INCREMENT KG/SU CM	MAXIMUM SHEAR STRESS KG/SQ CM	SHEAR STRESS RATIU	LOAD HATE KG/SQ CM PER HOUR
443	121	(3)	14)	151	161	171	[4]	(9)	(10)	1111	1121	1131
127	1	0.91	100.	1.0	0.087	2770.	4313.	513 513	0.105	0.637	0.1641	17.142
130	i	U. 85	100.	2.5	J. 087	4800.	6643.	476	0.213	1.067	0.1497	7.759

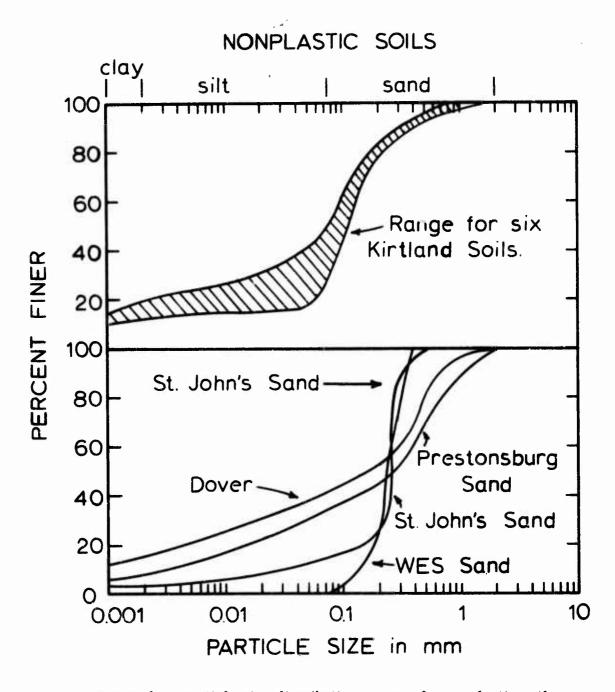


Figure 10. Particle size distribution curves for nonplastic soils.

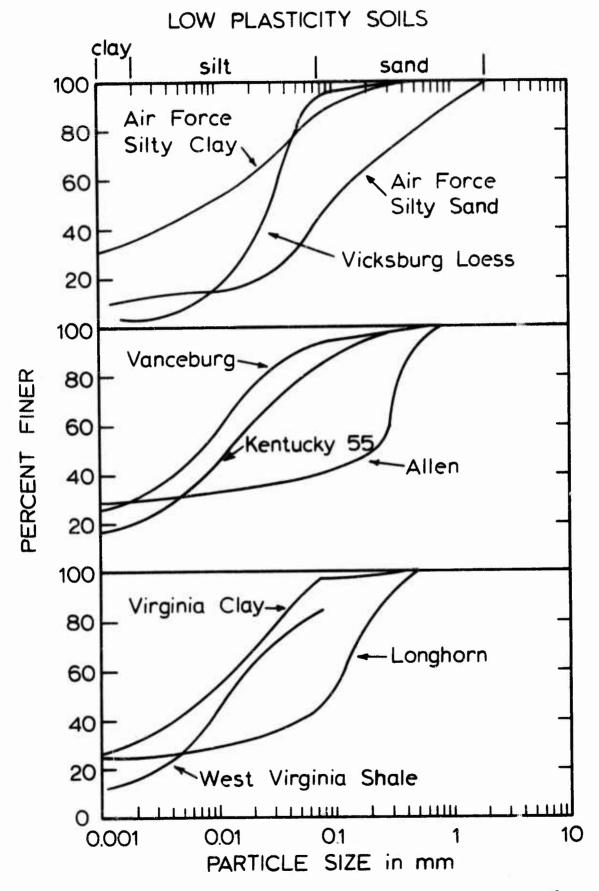


Figure 11. Particle size distribution curves for low plasticity soils.

HIGH PLASTICITY SOILS

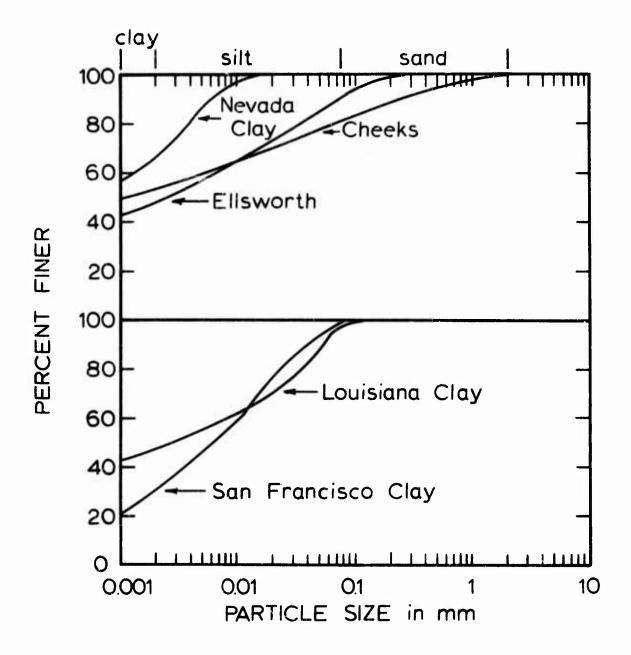


Figure 12. Particle size distribution curves for high plasticity soils.

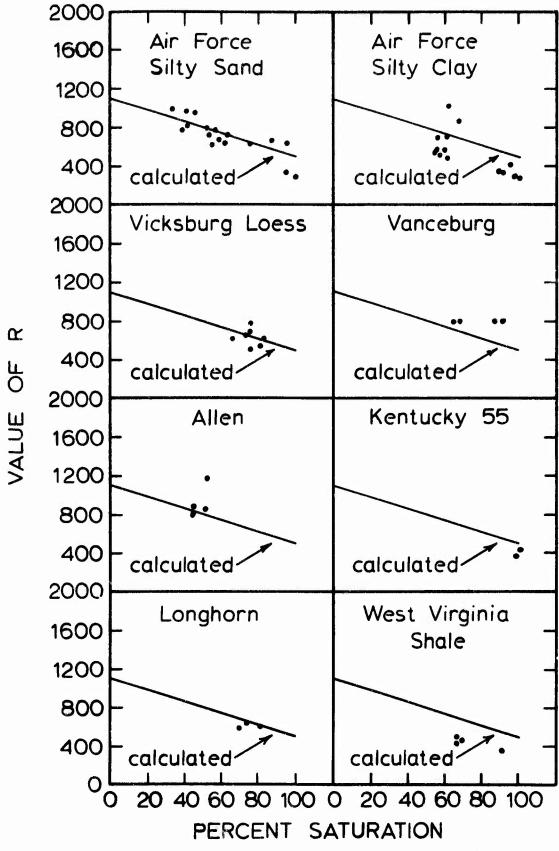


Figure 13. Comparison of the variation of experimental and calculated values of R with percent saturation, Air Force Silty Sand, Air Force Silty Clay, Vicksburg Loess, Vanceburg, Allen, Kentucky 55, Longhorn, and West Virginia Shale.

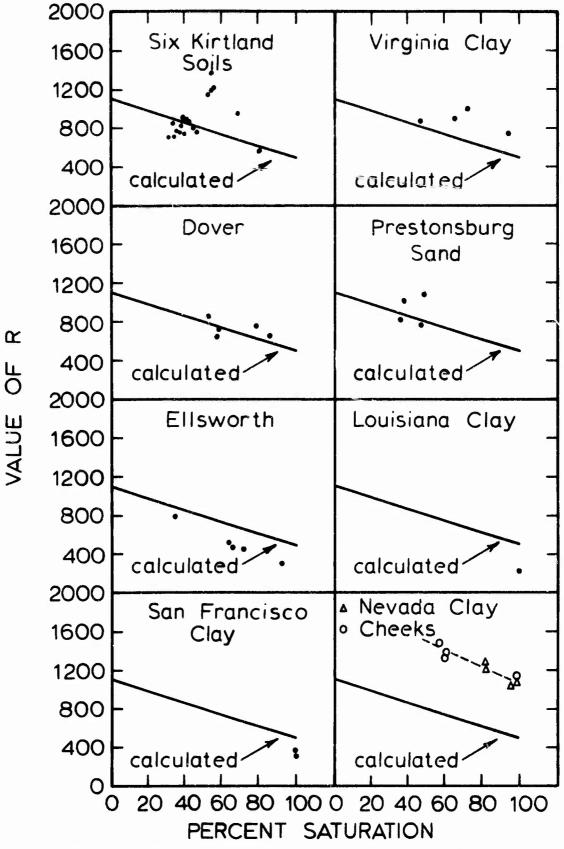


Figure 14. Comparison of the variation of experimental and calculated values of R with percent saturation, Six Kirtland Soils, Virginia Clay, Dover, Prestonsburg Sand, Ellsworth, Louisiana Clay. San Francisco Clay, Cheeks, and Nevada Clay.

experimental values. The weakest comparison is with two of the high plasticity soils, the Cheeks and Nevada Clays. Here the measured and calculated values differ by a factor of 2. However, the Louisiana and San Francisco Clays are also high plasticity soils, and they more nearly conform to the pattern of the other cohesive soils than to the Cheeks and Nevada Clays. Hence, the use of equation 3 for high plasticity soils is rather uncertain but is the best available at present.

shown in table 5. The average value from 10 tests of dry WES sand is 1102 with about 10 percent scatter. The average value for the St. John's Sand with tests at both 88 and 100 percents saturation is 1224, with the value for test no. 76 being excessively high. The average for the St. John's Sand excluding test no. 76 is 1104, which nearly equals the value for the dry sand. From this it was concluded that for sands with less than 15 percent fines R = 1100 could be used for all percents saturation, as shown by the first of equation 3.

The value of R defines the relationship between τ_{max} and G_{max} . As shown in Appendix I, equation 11 can be rearranged to eliminate τ_{max} for reference strain, and obtain the reference strain in terms of G_{max} as given by equation 1. Therefore, the data in figures 13 and 14 and in table 5 defining the value of R, are the basis for the equations given in Section II, paragraph 2 for determination of the reference strain, and indicate the degree of accuracy that may be expected when using that part of the practical procedure.

Table 5. Values of R for Sands with Less than 15 Percent Fines.

WES Sa	und (dry)	St.	John's Sand	
Test No.	Value of R	Test No.	Percent Saturation	Value of R
34	1045	76	100	1584
35	1094	77	100	1313
36	1046	79	88	1064
37	1026	80	88	934
38	1101			
39	1116			
83	990			
89	1167			
101	1125			
112	1311			
Avg.	1102			1224

3. Comparison of Shear Modulus Values

The variations of normalized shear modulus with normalized strain, as given by equations 8 and 9, are shown in figure 15 for different values of a. When a=0 the relationship is hyperbolic. Figure 15 shows that higher values of a give lower values of G/G_{max} for a given γ/γ_r . Using measured values for G, G_{max} , τ_{max} , and γ , experimental values of the normalized shear modulus and normalized strain can be calculated. In figures 16 through 22 such experimental values are compared to the values given by equations 8, 9, and 10 of the practical procedure. Figures 16 through 22 are for the first cycle of loading. One test of each different soil was chosen to show the comparison. The tests shown were not chosen because they gave the best comparisons. The parameters necessary for obtaining the calculated curve from equations 8, 9, and 10 are shown in the upper right hand corner of each graph. The shapes of the calculated curves correspond to various values of a between 0.33 and 5.34. The accuracy of the practical procedure is quite good considering the range of soil types and conditions to which they are applied.

Comparisons for the 10th and 100th cycles are shown in figures 23 and 24, respectively. For these comparisons the measured value of normalized strain along with the value of a calculated from equation 10 were used to calculate γ_h from equation 9. This value of γ_h was plotted versus the measured normalized shear modulus, thus locating the different symbols in figures 23 and 24 representing the measured relationship. The calculated relationship shown by the curves in figures 23 and 24 is given by equation 8 and is hyperbolic,

Figure 15. Variation of normalized shear modulus with normalized strain as given by the practical procedure for various values of a.

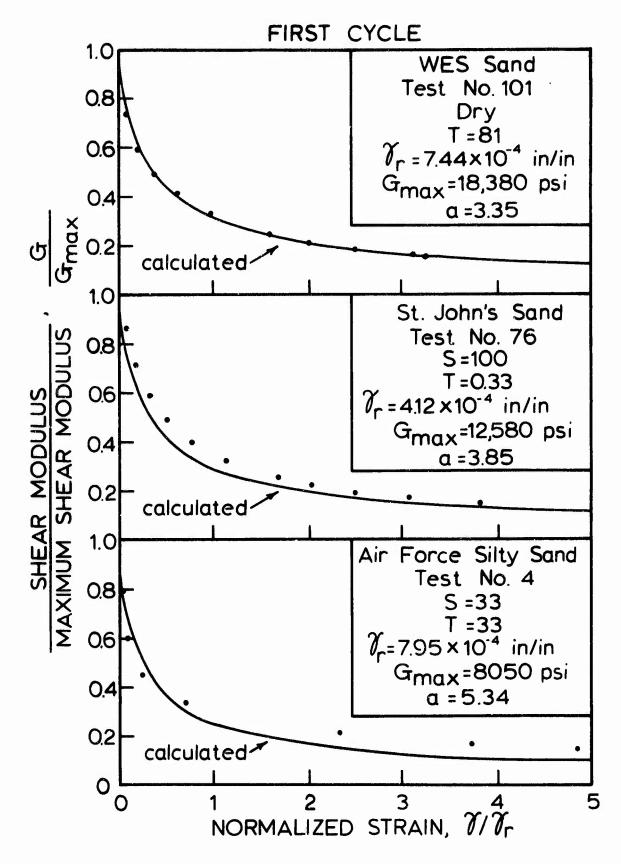


Figure 16. Comparison of measured and calculated values of shear modulus, first cycle, WES Sand, St. John's Sand, and Air Force Silty Sand.

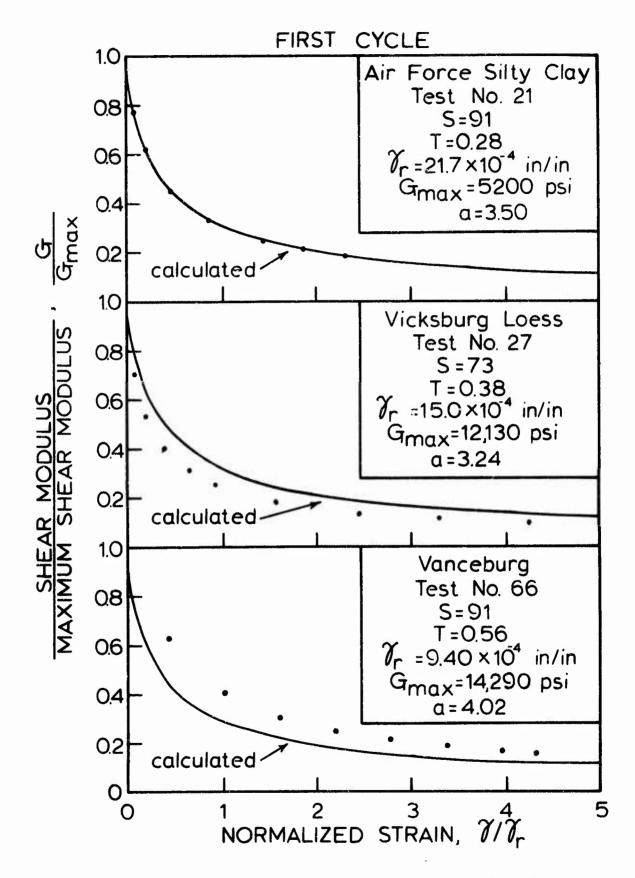


Figure 17. Comparison of measured and calculated values of shear modulus, first cycle, Air Force Silty Clay, Vicksburg Loess, and Vanceburg.

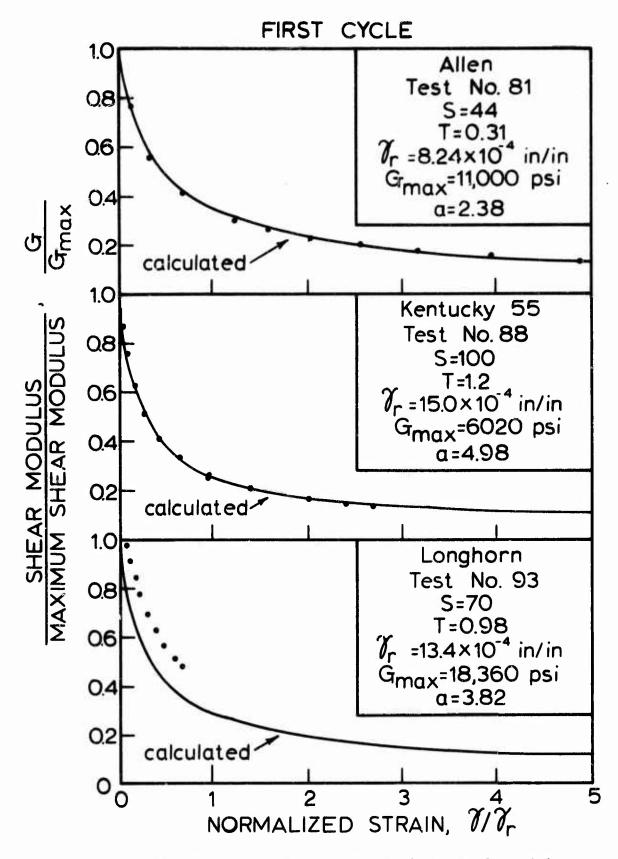


Figure 18. Comparison of measured and calculated values of shear modulus, first cycle, Allen, Kentucky 55, and Longhorn.

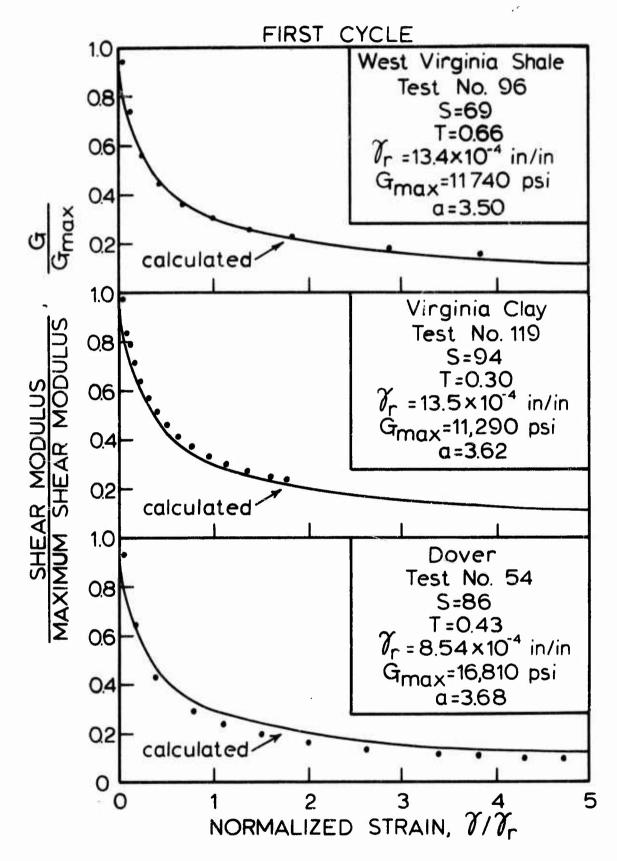


Figure 19. Comparison of measured and calculated values of shear modulus, first cycle, West Virginia Shale, Virginia Clay, and Dover.

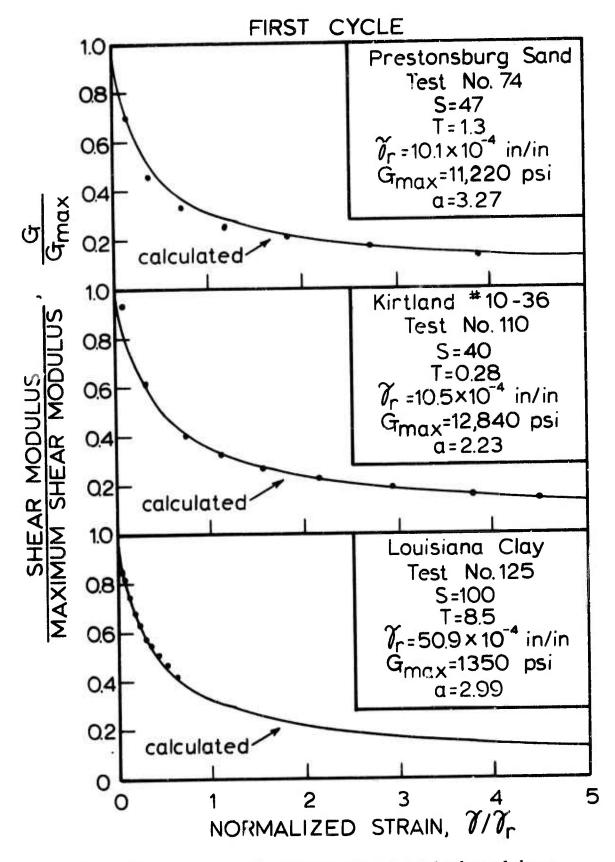


Figure 20. Comparison of measured and calculated values of shear modulus, first cycle, Prestonsburg Sand, Kirtland #10-36, and Louisiana Clay.

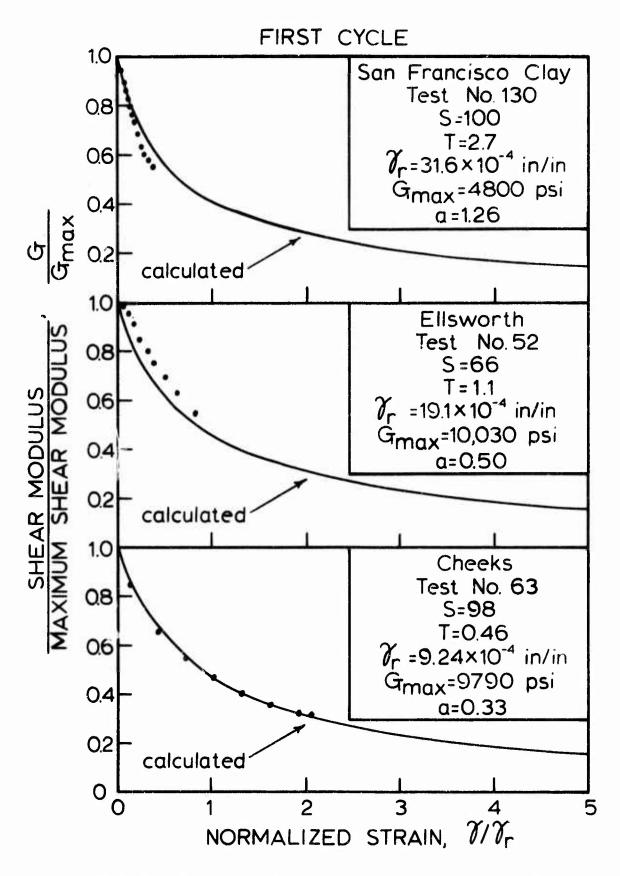


Figure 21. Comparison of measured and calculated values of shear modulus, first cycle, San Francisco Clay, Ellsworth, and Cheeks.

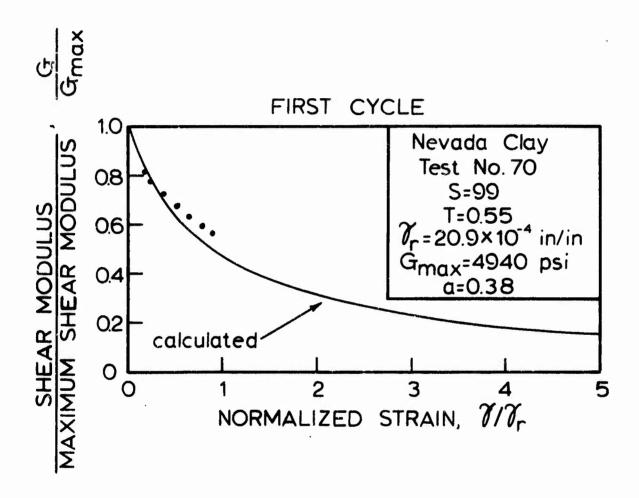


Figure 22. Comparison of measured and calculated values of shear modulus, first cycle, Nevada Clay.

since the hyperbolic strain was used as abscissa. The comparison between measured and calculated values for the 10th and 100th cycles is good.

The hyperbolic strain is used in figures 23 and 24 so that data for different soils with different values of S and T can all be shown on the same graph. When G/G_{max} is plotted versus γ/γ_r , the expected relationship depends on the values of N, S, T and soil type. A different calculated curve would be needed for each test, as was done in figures 16 through 22 for N = 1.

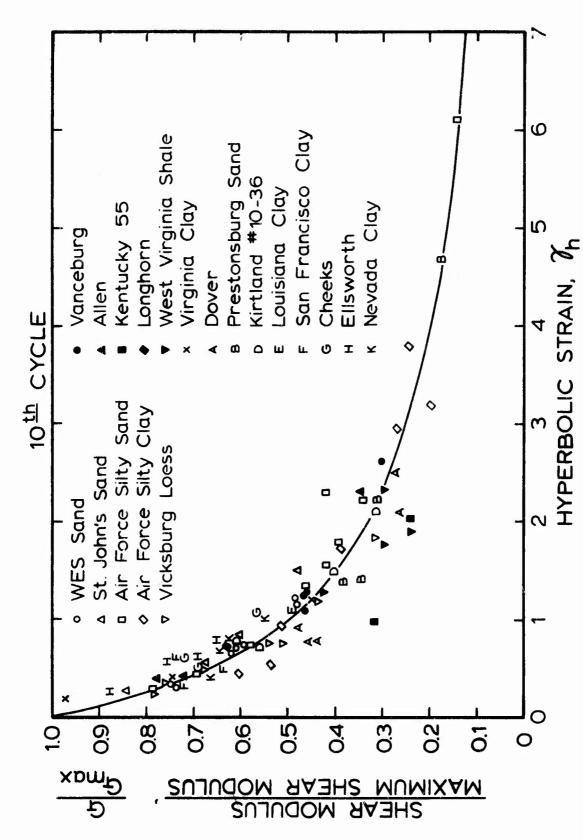


Figure 23. Comparison of measured and calculated values of shear modulus, 10th cycle.

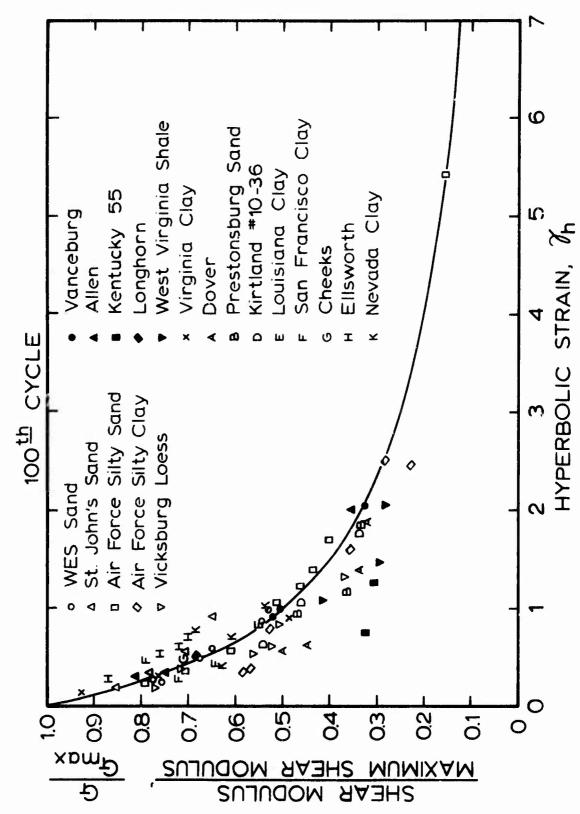


Figure 24. Comparison of measured and calculated values of shear modulus, 100th cycle.

SECTION IV

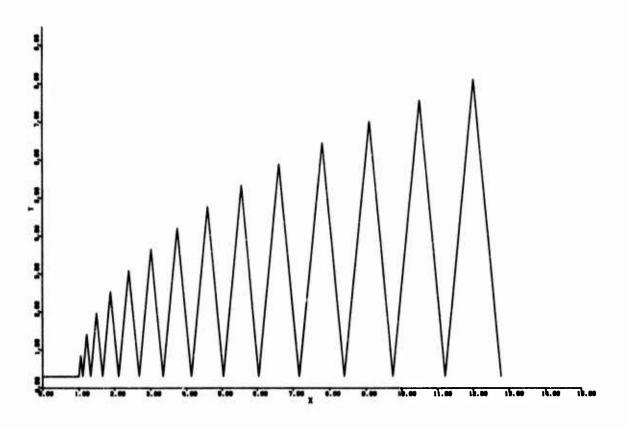
MIXED LOADING AMPLITUDES AND REST PERIODS

1. Objective

For the majority of tests discussed herein a cyclic shear loading was applied continuously, where the next cycle began immediately as the preceding cycle ended, and the loading amplitude was constant for all cycles as was the rate of loading and unloading. Because actual air traffic involves varying length rest periods between load applications and a mixture of light and heavy loadings, it was desirable to assess these effects by conducting tests where the loading could be programmed. Then a soil specimen could be subjected to rest periods, with zero load, between cycles of loading and the amplitude of loading could be different for successive cycles of loading.

2. Loading Programs and Recorded Stress-Strain Relations.

Three primary loading sequences were used for mixed amplitude studies. The first was an increasing load sequence as shown in the top of figure 25. In this figure the ordinate is proportional to the applied shear stress and the abscissa is time. The slope of the lines in this figure are proportional to the rate of loading. The loading program in the top of figure 25 produces a sequence of loads with monotonically increasing amplitudes but with a constant rate of loading and unloading. The second loading sequence shown in the bottom of figure 25 produces a large first load with subsequent loads decreasing in amplitude monotonically. The third sequence was a combination of these two where the sample was subjected first to the increasing load sequence followed by the



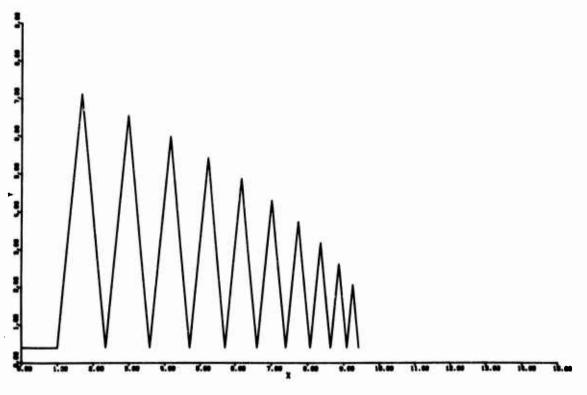


Figure 25. Loading Program.

decreasing load sequence.

Examples of the recorded stress-strain relations for the increasing load sequence are shown in figures 26, 27, and 28 for silty clay, silty sand, and WES loess specimens, respectively. Figure 26a shows the results of applying 46 cycles of increasing load to a specimen of silty clay. These 46 cycles were applied continuously without a rest period. At the end of the 46th cycle there was a short rest period while the recording paper was changed. After the rest period the increasing load sequence was continued and the results of cycles 47 through 67 were recorded in figure 26b. The recording was approaching the top of the paper on cycle 63 necessitating a change in stress scale during this cycle. The scales corresponding to various sections of the recording are shown on the figure. At the end of cycle 67 there was a short rest period as the recording paper was changed. The increasing load sequence was again continued and the results for cycles 68 through 83 are shown in figure 26c. At the end of the 82nd cycle the pen of the recorder was shifted electronically to the left and up because the strain was becoming large with the recording approaching the edge of the paper. The loading curve for cycle 83 is shown above the main recording and as can be seen failure of the specimen occurred on the 83rd cycle. Similar procedures were used for the recordings in figures 27 and 28.

The result of applying the decreasing load sequence shown in the bottom of figure 25 to a specimen of silty sand is shown in the top of figure 29. This same decreasing load sequence was applied to a specimen of silty clay producing the results shown in the bottom of figure 29. However, the specimen of silty

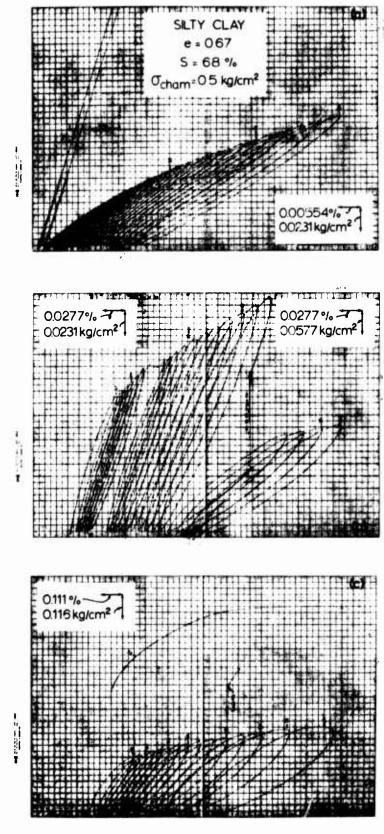
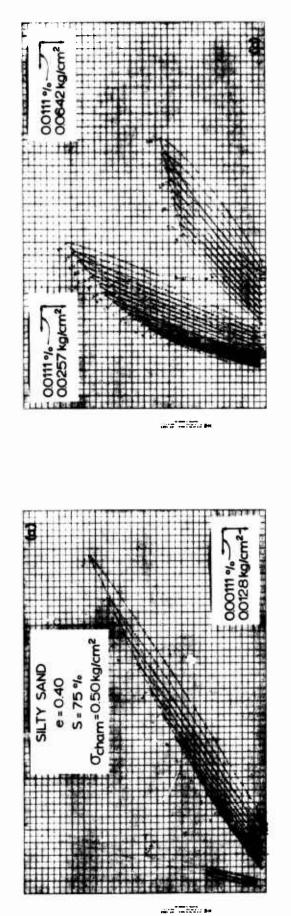
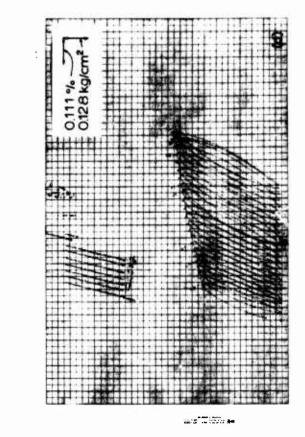


Figure 26. Recorded stress-strain relation, increasing load sequence, silty clay.





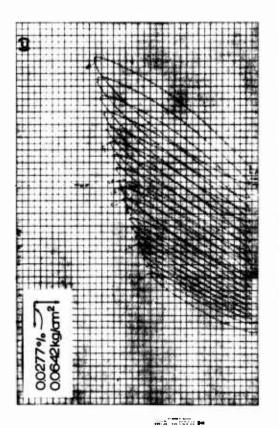


Figure 27. Recorded stress-strain relation, increasing load sequence, silty sand.

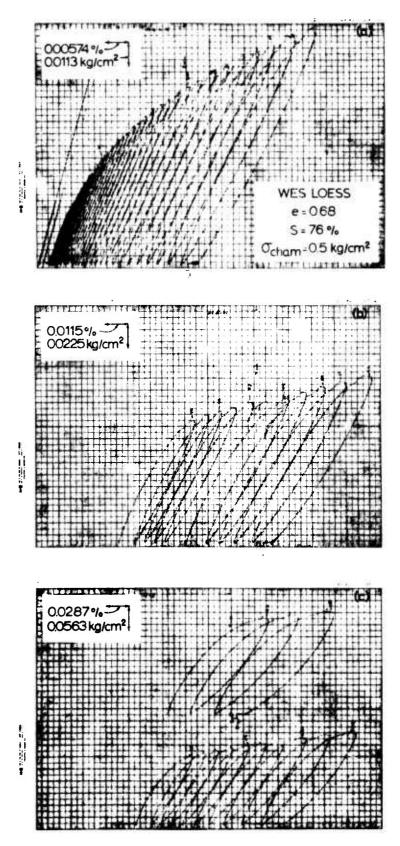
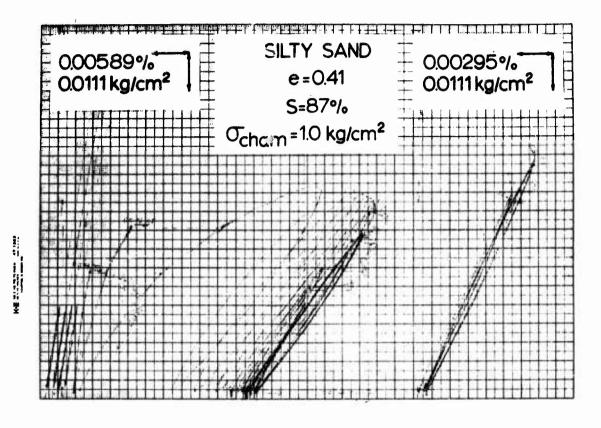


Figure 28. Recorded stress-strain relation, increasing load sequence, Vicksburg Loess.



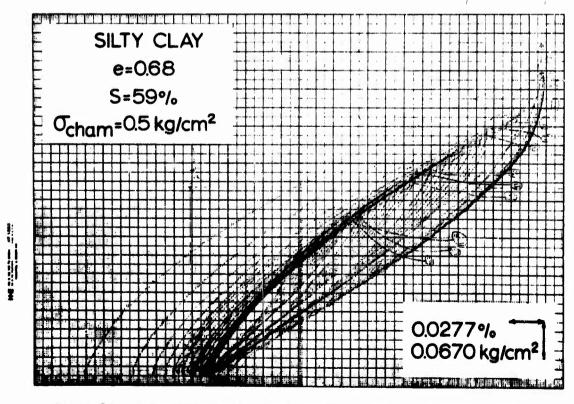


Figure 29. Recorded stress-strain relation, decreasing load sequence.

clay had previously been subjected to the increasing load sequence in the top of figure 25. From these results it can be seen that for decreasing load sequences there is little additional accumulation of set strain as the smaller and smaller loads are applied. Hence, the curves for successive loads trace over the same area of the recording.

3. Effect on the Shear Modulus

The values of normalized modulus for all of the mixed amplitude tests are plotted versus normalized strain for the silty clay in figure 30 and for the silty sand in figure 31. The open symbols are for increasing amplitude loading sequences and the solid symbols are for decreasing amplitude loading sequences. Also shown on these figures are the curves calculated from equations 8 and 9 for values of a = 0 and a = 2. As shown in figure 17, the value of a for the first cycle of test no. 21 on the Air Force silty clay is 3.50. This value decreased with number of cycles and approaches zero for a large number of cycles. For the first cycle of test no. 4 on the Air Force silty sand in figure 16 the value of a = 5.34. Again this value approaches zero for a large number of cycles. Since most of the values shown in figures 30 and 31 fall in the range between a = 0 and a = 2, it appears that the effects of the mixed amplitudes and rest periods on the shear modulus are smaller than the scatter in measured values from test to test of the same soil. Therefore, it is concluded that for practical purposes the procedure outlined in Section II can be applied to mixed traffic conditions.

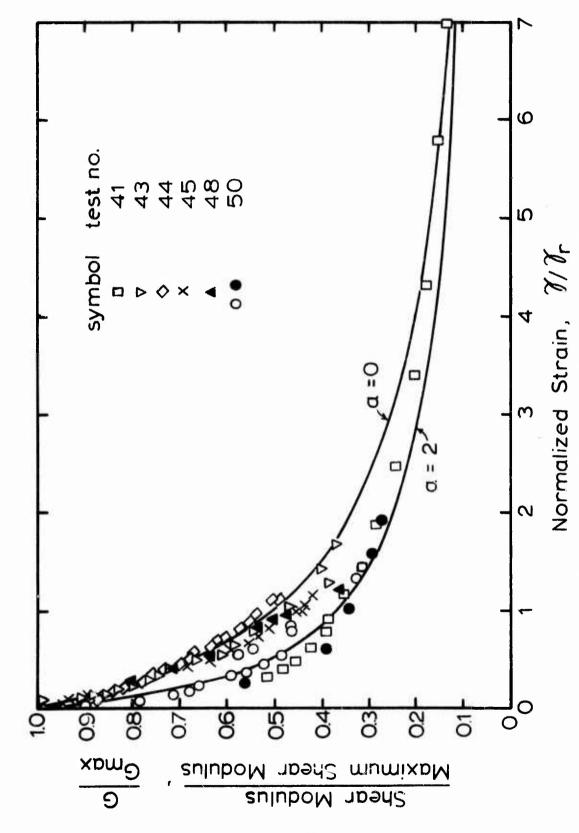


Figure 30. Normalized shear modulus versus normalized strain for silty clay, mixed amplitudes and rest periods.

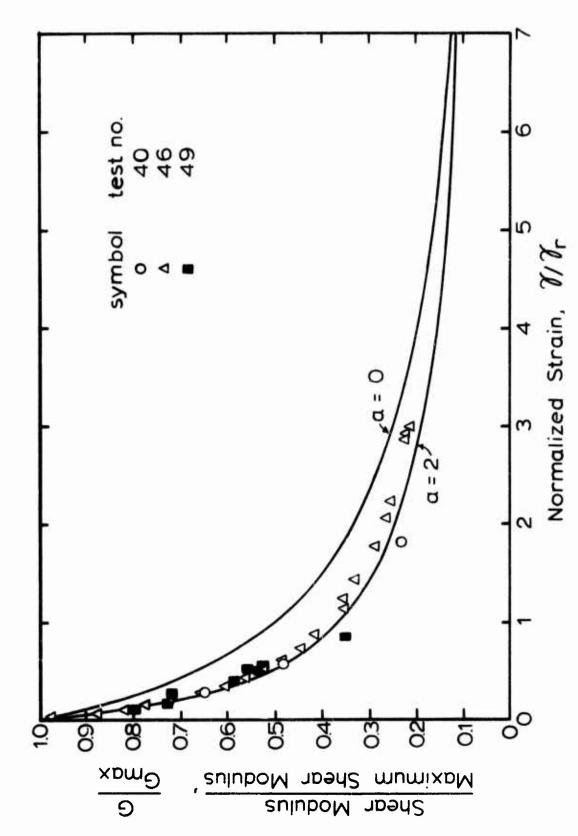


Figure 31. Normalized shear modulus versus normalized strain for silty sand, mixed amplitudes and rest periods.

SECTION V

CONCLUSIONS

- 1. A practical procedure for reducing the shear modulus of soils with increasing strain amplitude has been developed. It has been shown that the procedure gives reasonably accurate results compared to values measured in the laboratory, for a wide variety of soil types and conditions. The study of mixed amplitudes and rest periods indicates that the procedure can also be applied to mixed traffic conditions.
- 2. The shear modulus of soils in general varies between extreme limits. It may commonly be less than 1000 psi or greater than 50000 psi. Likewise, the relationship between shear modulus and strain level is extremely variable. The picture can be simplified by normalization. The shear modulus is normalized by considering G/G_{max} , but the relationship between normalized shear modulus and strain level is still quite variable; because, a given strain does not have the same effect on all soils or on the same soil under different states of stress. When strain is also normalized by dividing by the reference strain, $\gamma_r = \tau_{max}/\sigma_{max}$, a single relationship for all soils between normalized shear modulus and normalized strain is approached. However, this relationship is still affected somewhat by soil type, percent saturation, and number of cycles and rate of loading. A single relationship for all soils and conditions is finally obtained by defining the hyperbolic strain that depends mainly on the normalized strain but accounts for the residual effects of soil type, S, N, and T. Hence, these concepts of reference strain and hyperbolic strain have made possible the

development of the practical procedure of paragraph 1.

- 3. The 123 simple shear tests of 24 different soils reported herein show the following parameter effects on the relationship between normalized shear modulus and normalized strain. For a given value of normalized strain, the normalized shear modulus increased with number of cycles of loading, decreased with increasing percent saturation, and increases with rate of loading. There appears to be very little effect of void ratio on this relationship.
- 4. The reference strain that is so important to the procedure of paragraph 1 depends on $\tau_{\rm max}$ and $G_{\rm max}$. For pavement evaluation the value of $G_{\rm max}$ is to be determined by the nondestructive vibratory test. It is not desirable to have to measure the value of $\tau_{\rm max}$. In this study a relationship between reference strain and $G_{\rm max}$ has been established, eliminating the necessity for estimating $\tau_{\rm max}$. The relationship depends on void ratio, percent saturation and plasticity index of the soil. It is best defined for nonplastic and low plasticity soils, where the data in figures 13 and 14 indicate that the error is less than 25 percent for about 50 percent of the cases. It should be remembered that a 25 percent error in the value of $\gamma_{\rm r}$ does not produce a correspondingly large error in the determination of G. For a value of normalized strain of one, a 25 percent error in reference strain produces about 12 percent error in the determination of G. The relationship is much less well defined for high plasticity soils with liquid limit greater than 50, where the error in determination of reference strain may be greater than 100 percent.
 - 5. Assuming the normalized strain is accurately known, the error to be

expected in the relationship between normalized strain and normalized shear modulus is shown by the data in figures 16 through 24, where the vast majority of points fall within plus or minus 20 percent.

6. Finally, an electromagnetic, hollow cylinder, simple shear test capable of accurate measurement of the stress-strain relations for soils over a wide range of strains, from about 10⁻⁵ in/in to strains corresponding to failure, and where the loading can be programmed to test effects of mixed amplitudes and rest periods has been developed.

APPENDIX I

DERIVATION OF REFERENCE STRAIN EQUATION

Neglecting the effects of overconsolidation, for a simple shear state of stress

$$\tau_{\max} = \bar{\sigma}_{0} \sin \bar{\phi} \tag{12}$$

where $\overline{\sigma}_0$ = effective mean principal stress and $\overline{\phi}$ = effective angle of shearing resistance. Also as stated in reference 1,

$$G_{\text{max}} = 1230 \text{ F } \bar{\sigma}_0^{1/2}$$
 (13)

Equation 13 is based on many tests of saturated soils. Solving for $\bar{\sigma}_0$ in equation 13, substituting into equation 12 and replacing the factor 1230 by R, since this factor may not be constant for partially saturated soils

$$\tau_{\text{max}} = \frac{G^2_{\text{max}} \sin \bar{\phi}}{R^2 F^2} \tag{14}$$

As shown in reference 2, the effective angle of shearing resistance can be defined approximately by

$$\sin \bar{\phi} = 0.6 - 0.25 \text{ (PI)}^{0.6}$$
 (15)

Substituting equation 15 into equation 14

$$\tau_{\text{max}} = \frac{G^2_{\text{max}}}{R^2 F^2} \quad [0.6 - 0.25 \text{ (PI)}^{0.6}] \tag{16}$$

The reference strain

$$\gamma_{r} = \frac{\tau_{\text{max}}}{G_{\text{max}}} \tag{17}$$

^{2.} Lambe, T. W. and Whitman, R. V., Soil Mechanics, Wiley, 1969, p. 307.

Substituting equation 16 into equation 17

$$\gamma_{\mathbf{r}} = \frac{G_{\text{max}}}{R^2 F^2} = [0.6 - 0.25 \text{ (PI)}^{0.6}]$$
 (1)

APPENDIX II

TESTING METHODS AND PROCEDURES

A simple shear test, where a hollow cylinder of soil was confined in a pressure chamber and loaded torsionally about the axis of the cylinder, was employed for this research. The torsional load was applied electromagnetically by requiating the voltage to four large coils within the fields of four corresponding permanent magnets. The system was capable of applying a maximum torque of approximately 60 kg-cm. The voltage was produced by a wide frequency range, triangular wave signal generator, amplified by a 50 watt DC amplifier. With this electromagnetic system the rate and amplitude of loading could be changed by simply turning the appropriate knob on the signal generator. The rate of loading for the tests varied from approximately 0.2 to 450 kg/cm per hr. Inertial forces were negligible for these rates of loading. Torque and angular motion were measured with electrical transducers located inside the pressure chamber, thus eliminating the influence of friction and apparates deformation on the measurements. The signals from these transducers were recorded with an X-Y recorder.

Most of the soil specimens tested were subjected to approximately 1000 cycles of a constant-amplitude shear stress. Examples of the recorded shear stress-strain relations are shown in figure 32. This figure illustrates the range of strain amplitudes tested. For test no. 7, top of figure 32, the cyclic shear stress was approximately 80 percent of the strength of the sample. For test no. 15, record no. 1, bottom of figure 32, the cyclic shear stress was only 2.5

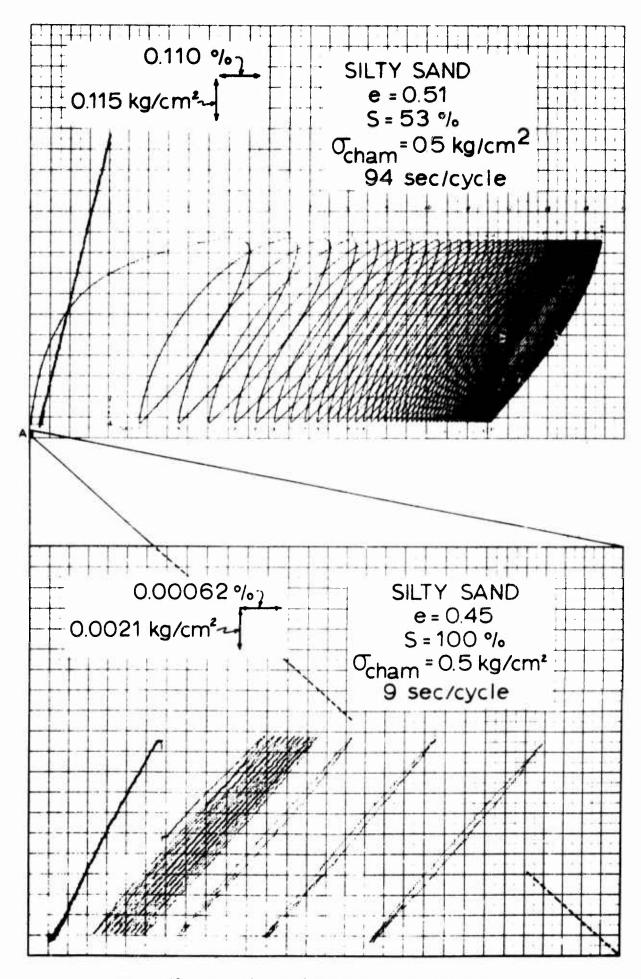


Figure 32. Typical recorded stress-strain relations.

percent of the strength of the sample. The entire graph sheet for test no. 15, if changed to the scale of test no. 7, would fit in the small black rectangle at A in the lower left corner of test no. 7. Cyclic loading tests were conducted for strain amplitudes as small as about 10^{-5} in/in to strain amplitudes as large as 0.5×10^{-2} in/in. After the cyclic loading of each sample, the load was increased to failure in order to measure $\tau_{\rm max}$.